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**COMPUTER PROGRAM FOR CALCULATING FLOW
DISTRIBUTION IN A RADIAL-INFLOW TURBINE**

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SUMMARY

A FORTRAN computer program for flow analysis of a radial inflow gas turbine is given. The program obtains a meridional solution on the mean surface between the blades, followed by solutions on hub, mean, and shroud blade-to-blade surfaces, in a single computer run. Suggestions for modifying the program for use with other types of turbomachines are given. Techniques for overcoming convergence problems are discussed.

INTRODUCTION

A method of flow analysis for any turbomachine is summarized in reference 1, with the details being presented in references 2 and 3. Reference 2 gives the method applied to the meridional plane. Reference 3 gives the method for obtaining a blade-to-blade solution, using information obtained from the meridional plane solution. The FORTRAN program for each solution as applied to a radial inflow gas turbine is given in the appropriate reference.

As stated in reference 3, a FORTRAN program has been written which combines both programs. The combined program obtains first the meridional solution, followed by three blade-to-blade solutions (hub, mean, and shroud), in a single computer run. Either the meridional solution or blade-to-blade solution may be obtained separately.

This report presents this FORTRAN program, with sample output for the example rotor of reference 1. Also suggestions are given for modifying the program for other types of turbomachines, and for overcoming convergence problems which may be encountered.

PROGRAM VARIABLES

The variable names used in the combined program are the same as listed in reference 2 and 3 with the addition of the following variables.

THH(I)	θ	coordinate of mean blade surface at I th quasi-orthogonal along hub
THM(I)	θ	coordinate of mean blade surface at I th quasi-orthogonal along mean streamline

THS(I)	θ	coordinate of mean blade surface at I^{th} quasi-orthogonal along shroud
PION	π/N	
TPION	$2\pi/N$	
THHI(I)	θ	coordinate of suction surface of complete blade at I^{th} quasi-orthogonal along hub
THMI(I)	θ	coordinate of suction surface of complete blade at I^{th} quasi-orthogonal along mean blade-to-blade stream surface
THSI(I)	θ	coordinate of suction surface of complete blade at I^{th} quasi-orthogonal along shroud
THHKH(I)	θ	coordinate of pressure surface of splitter blade at I^{th} quasi-orthogonal along hub
THMKH(I)	θ	coordinate of pressure surface of splitter blade at I^{th} quasi-orthogonal along mean blade-to-blade stream surface
THSKH(I)	θ	coordinate of pressure surface of splitter blade at I^{th} quasi-orthogonal along shroud
THHKP(I)	θ	coordinate of suction surface of splitter blade at I^{th} quasi-orthogonal along hub
THMKP(I)	θ	coordinate of suction surface of splitter blade at I^{th} quasi-orthogonal along mean blade-to-blade stream surface
THSKP(I)	θ	coordinate of suction surface of splitter blade at I^{th} quasi-orthogonal along shroud
THHKMX(I)	θ	coordinate of pressure surface of complete blade at I^{th} quasi-orthogonal along hub
THMKMX(I)	θ	coordinate of pressure surface of complete blade at I^{th} quasi-orthogonal along mean blade-to-blade stream surface
THSKMX(I)	θ	coordinate of pressure surface of complete blade at I^{th} quasi-orthogonal along shroud

MODIFICATIONS REQUIRED FOR OTHER APPLICATIONS

The program as written can be used for a radial inflow gas turbine. It should be emphasized that the listing of the combined program is being published as a guide to the technique used in programming the quasi-

orthogonal method of flow analysis. If it is desired to obtain a program for analysing any type of turbomachine other than a radial inflow gas turbine, several modifications as indicated below will have to be made to the program. This will require a reasonable understanding of the program and the equations involved. With this understanding, the required modifications can be easily made. Modifications which are necessary for some particular cases are listed below:

(1) Changes required for a compressor are:

(a) Rotation of coordinates for use in calculating spline fit curve for the meridional streamlines. It is assumed that the inlet is to the left with the positive direction to the right, and that the quasi-orthogonals are numbered starting with 1 at the inlet. The coordinates for the streamlines must be rotated 45° in the direction opposite to that in the present program. The modifications needed are to statements for calculating AB(I) and AC(I) near statements 160 and 390, and for calculating AL(I,K) four statements after statement 160.

(b) The sign of CEF as calculated just before statement 150 must be reversed.

(2) Changes required for a pump or liquid turbine.

Statements involving temperature or density must be modified to eliminate temperature, and to allow for constant density.

(3) Changes required for axial flow machines.

Statements for rotating streamline coordinates for use in calculating the spline fit curve for the meridional streamlines should be modified to eliminate this rotation. The statements involved are the same as those mentioned under 1a above.

CONVERGENCE

It may be found that with some geometries there is a convergence problem. This is helped by using as few and as evenly spaced quasi-orthogonals as possible, consistent with a reasonable accuracy. The number of streamlines used does not appreciably affect the number of iterations. Of course, it does affect the computer time required per iteration. Another factor which helps convergence is to reduce the streamline correction factor, CORFAC, (see ref. (2), p. 23 and fig. 5). Of course, this reduces the rate of convergence, but may result in convergence when a solution could not be obtained otherwise.

However, in some cases when the initial streamline estimate is poor it may be necessary to make the streamline correction factor, CORFAC, so small that the convergence rate is too slow. One way of improving the initial

streamline estimate is to use a large streamline correction for the first iteration, say CORFAC = 1. This will bring the streamlines fairly close to the correct position, but will result in rather uneven streamlines. If the streamlines are then smoothed sufficiently, the convergence will be generally much better than when working from the original position. It has not been found helpful to make this large streamline correction more than once. There are several mathematical techniques for smoothing (ref. 4).

Another problem that may be encountered with compressible fluids is that calculations may indicate choking weight flow less than the desired weight flow in the early iterations. This usually leads to a problem in convergence. However, this may be overcome by obtaining a solution based on a slightly lower weight flow, followed by a solution based on the desired weight flow. Possibly more than one increment of weight flow increase will be required to increase the flow up to the desired weight, or perhaps the desired weight flow will prove to be actually in excess of the choking weight flow.

PROGRAM LISTING

SIBFTC FIXED DECK

SUBROUTINE FIXED(HUB,MEAN,SHROUD)

C CALCULATION OF VELOCITY AND PRESSURE DISTRIBUTION IN A RADIAL FLOW
C TURBO MACHINE
C

```

C      COMMON SRW,KMX ,MX,WT,XN,GAM,AR,TYPE,BCDP,TEMP,ALM,RHO,TOLER,
1      PLOSS,NPRT,ITER1,BETIN,WTOLER,T-H1,THHKH,THHKP,THHKMX,THM1,
2      THMKH,THMKP,THMKMX,THS1,THSKH,THSKP,THSKMX,Z ,R ,AB,AC,AD,
3      RUNO,MXBL,PIOV,W,BETA1,WTR,CURV
      DIMENSION AL(21,21),BETA(21,21),CAL(21,21),CBFTA(21,21),
      ICURV(21,21),DN(21,21),PRS(21,21),R(21,21),Z(21,21),SM(21,21),
      2SA(21,21),SB(21,21),SC(21,21),SD(21,21),SAL(21,21),SBETA(21,21),
      ATN(21,21),TT(21,21),WA(21,21),WTR(21,21)
      DIMENSION AB(22),AC(22),AD(22),RA(21),DELBTA(21),ORDM(21),
      10TDR(21),OTDZ(21),OWMDM(21),OWTDM(21),RH(21),RS(21),7H(21),ZS(21),
      2THTA(21),WTL(21),XR(21),XT(21),XZ(21),BETA1(3),AA(3)
      DIMENSION THH(21),THM(21),THS(21),THH1(21),THHKH(21),THHKP(21),
      1 THHKMX(21),THM1(21),THMKH(21),THMKP(21),THMKMX(21),THS1(21),
      2 THSKH(21),THSKP(21),THSKMX(21)
      INTEGER RUNO,TYPE,BCDP,SRW,HUB,SHROUD
      RUNO=0
10 READ (5,1010)MX,KMX,MR,MZ,W,WT,XN,GAM,AR

      ITNO = 1
      RUNO=RUNO+1
      WRITE (6,1020) RUNO
      WRITE (6,1010)MX,KM:MR,MZ,W,WT,XN,GAM,AR
      READ (5,1010)TYPE,BCDP,SRW,MXBL,TEMP,ALM,RHO,TOLER,PLOSS,WTOLER

      WRITE(6,1010)TYPE,BCDP,SRW,MXBL,TEMP,ALM,RHO,TOLER,PLOSS,WTOLER
      READ (5,1010)MHTA,VPRT,ITER,NULL,SFACT,ZSPLIT,BETIN,CORFAC

      WRITE(6,1010)MHTA,VPRT,ITER,NULL,SFACT,ZSPLIT,BETIN,CORFAC
      ITER1 = ITER
      READ(5,1030){ZS(I),I=1,MX}
      WRITE(6,1030){ZS(I),I=1,MX}
      READ(5,1030){ZH(I),I=1,MX}
      WRITE(6,1030){ZH(I),I=1,MX}
      READ(5,1030){RS(I),I=1,MX}
      WRITE(6,1030){RS(I),I=1,MX}
      READ(5,1030){RH(I),I=1,MX}
      WRITE(6,1030){RH(I),I=1,MX}
      DO 20 I=1,MX
      ZS(I)=ZS(I)/12.
      ZH(I)=ZH(I)/12.
      RS(I)=RS(I)/12.
20 RH(I)=RH(I)/12.
      IF(TYPE.NE.0) GO TO 40
      WA(1,1) = WT/RHO/(ZS(1)-Z4(1))/3.14/(RS(1)+RH(1))
      DO 30 I=1,MX
      DN(I,KMX)=SQRT((ZS(I)-ZH(I))**2+(RS(I)-RH(I))**2)
      DO 30 K=1,KMX
      DN(I,K)=FLOAT(K-1)/FLOAT(KMX-1)*DN(I,KMX)
      WA(I,K)=WA(1,1)
      Z(I,K)=DN(I,K)/DN(I,KMX)*(ZS(I)-ZH(I))+ZH(I)

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30 R(I,K)=DN(I,K)/DN(I,KMX)*RS(I)-R(I)+R(I)
   GO TO 50
40 IF(TYPE.NE.1) GO TO 145
   CALL BCREAD(DN(1,1),DN(21,21))
   CALL BCREAD(WA(1,1),WA(21,21))
   CALL BCREAD(Z(1,1),Z(21,21))
   CALL BCREAD(R(1,1),R(21,21))
   WRITE(6,1040)
50 READ(5,1030)(THA(I),I=1,MTHTA)
   WRITE(6,1030)(THA(I),I=1,MTHTA)
   READ(5,1030)(XT(I),I=1,MTHTA)
   WRITE(6,1030)(XT(I),I=1,MTHTA)
   DO 60 K=1,MR
     READ(5,1030)(TN(I,K),I=1,MZ)
60 WRITE(6,1030)(TN(I,K),I=1,MZ)

   READ(5,1030)(XZ(I),I=1,MZ)
   WRITE(6,1030)(XZ(I),I=1,MZ)
   READ(5,1030)(XR(I),I=1,MR)
   WRITE(6,1030)(XR(I),I=1,MR)

C
C   END OF INPUT STATEMENTS
C
C   SCALING-CHANGE INCHES TO FEET AND PSI TO LB/SQ FT, INITIALIZE, CALCULATE
C   CONSTANTS
C
70 DO 90 K=1,MR
   DO 80 I=1,MZ
80 TN(I,K) = TN(I,K)/12.
90 XR(K) = XR(K)/12.
   DO 100 I=1,MZ
100 XZ(I) = XZ(I)/12.
   DO 110 K=1,KMX
110 SM(I,K)=0.
   BA(1)=0.
   DO 120 K=2,KMX
120 BA(K) = FLOAT(K-1)*WT/FLOAT(KMX-1)
   DO 130 I=1,MX
130 DN(I,1)=0.
   DO 140 I=1,MTHTA
140 XT(I)=XT(I)/12.
   ROOT = SQRT(2.0)
145 CONTINUE
   TOLER =TOLER/12.
   ZSPLIT = ZSPLIT/12.
   PLOSS=PLOSS*144.
   CI = SQRT(GAM*AR*TEMP)
   WRITE(6,1050) CI
   KMXM1 = KMX-1
   CP=AR*GAM/(GAM-1.)
   EXPOV = 1./(GAM-1.)
   BETIN = BETIN/57.29577
   ZINLET = (ZS(1)+ZH(1))/2.
   RINLET = (RS(1)+RH(1))/2.
   CALL LININT(ZINLET,RINLET,XZ,XR,TN,21,21,T)
   RB = RINLET*EXP(-.71*(2.+3.14159/(XN*SFAC)-T/RINLET))

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WRITE (6,1030) RB
GEF = -SIN(BETIN)/COS(BETIN)/RINLET/(RINLET-RB)**2
ERRDR=100000.

```

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C
C BEGINNING OF LOOP FOR ITERATIONS
C

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150 IF(ITER.EQ.0) WRITE (6,1060) ITNO
IF(ITER.EQ.0) WRITE (6,1070)
ERROR1=ERROR
ERROR=0.

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C
C START CALCULATION OF PARAMETERS
C

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DO 230 K=1,KMX
DO 160 I=1,MX
AB(I) = (Z(I,K)-R(I,K))/ROOT
160 AC(I)=(Z(I,K)+R(I,K))/ROOT
CALL SPLINE (AB,AC,MX,AL(1,K),CURV(1,K))
DO 170 I=1,MX
CURV(I,K)=CURV(I,K)/(1.+AL(I,K)**2)**1.5
AL(I,K) = ATAN(AL(I,K))-0.785398
CAL(I,K) = COS(AL(I,K))
170 SAL(I,K) = SIN(AL(I,K))
DO 180 I=2,MX
180 SM(I,K) = SM(I-1,K)+SQRT((Z(I,K)-Z(I-1,K))**2+(R(I,K)-R(I-1,K))**
1 2)
190 CALL SPLDER(XT(1),THTA(1),MTHTA,Z(1,K),MX,DTDZ(1))
DO 220 I=1,MX
T = 0.
IF(I.LE.MXBL) CALL LININT(Z(I,K),R(I,K),XZ,XR,TN,21,21,T)
IF(R(I,K).LE.RB)GO TO 200
DTDR(I)=GEF*(R(I,K)-RB)**2
GO TO 210
200 DTDR(I)=0.
210 TQ=R(I,K)*DTDR(I)
TP = R(I,K)*DTDZ(I)
TT(I,K)=T*SQRT(1.+TP*TP)
BETA(I,K)=ATAN(TP*CAL(I,K)+TQ*SAL(I,K))
SBETA(I,K) = SIN(BETA(I,K))
CBETA(I,K) = COS(BETA(I,K))
SA(I,K)=CBETA(I,K)**2*CAL(I,K)*CURV(I,K)-SBETA(I,K)**2/R(I,K)+
1SAL(I,K)*CBETA(I,K)*SBETA(I,K)*DTDR(I)
SC(I,K)=-SAL(I,K)*CBETA(I,K)**2*CURV(I,K)+SAL(I,K)*CBETA(I,K)
1*SBETA(I,K)*DTDZ(I)
AB(I)=WA(I,K)*CBETA(I,K)
220 AC(I)=WA(I,K)*SBETA(I,K)
CALL SPLINE(SM(1,K),AB,MX,DWMDM,AD)
CALL SPLINE(SM(1,K),AC,MX,DWDM,AD)
IF((ITER.LE.0).AND.(MOD(K-1,NPRT).EQ.0)) WRITE (6,1080) K
DO 230 I=1,MX
SB(I,K)=SAL(I,K)*CBETA(I,K)*DWMDM(I)-2.*W*SBETA(I,K)+DTDR(I)*
1R(I,K)*CBETA(I,K)*(DWDM(I)+2.*W*SAL(I,K))
SD(I,K)=CAL(I,K)*CBETA(I,K)*DWDM(I)+DTDZ(I)*
1R(I,K)*CBETA(I,K)*(DWDM(I)+2.*W*SAL(I,K))
IF((ITER.GT.0).OR.(MOD(K-1,NPRT).NE.0))GO TO 230
A= AL(I,K)*57.29577

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      B= SM(I,K)*12.
      E= TT(I,K)*12.
      G=BETA(I,K)*57.29577
      WRITE (6,1090) A,CURV(I,K),B,G,E, SA(I,K),SB(I,K),SC(I,K),SD(I,K)
230 CONTINUE
C
C END OF LOOP -- PARAMETER CALCULATION
C CALCULATE BLADE SURFACE VELOCITIES (AFTER CONVERGENCE)
C
      IF(ITER.NE.0) GO TO 260
      DO 250 K=1,K4X
      CALL SPLINE (SM(1,K),TT(1,K),MX,DELBT,AC)
      A=XN
      DO 240 I=1,MX
240 AB(I)=(R(I,K)*W+WA(I,K)*SBETA(I,K))*(6.283185*R(I,K)/ A-TT(I,K))
      CALL SPLINE (SM(1,K),AB,MX,DRDM,AC)
      IF (SFACT.LE. 1.0) GO TO 245
      A = SFACT*XN
      DO 244 I=1,MX
244 AB(I)=(R(I,K)*W+WA(I,K)*SBETA(I,K))*(6.283185*R(I,K)/ A-TT(I,K))
      CALL SPLINE (SM(1,K),AB,MX,AD ,AC)
245 DO 250 I=1,MX
      BETAD = BETA(I,K)-DELBT(I)/2.
      BETAT = BETAD+DELBT(I)
      COSBD = COS(BETAD)
      COSBT = COS(BETAT)
      IF(Z(I,K).LT.ZSPL(I)) DRDM(I) = AD(I)
      WTR(I,K)=COSBD*COSBT/(COSBD+COSBT)*(2.*WA(I,K)/COSBD+R(I,K)*W*
      1(BETAD-BETAT)/CBETA(I,K)**2+DRDM(I))
250 CONTINUE
C
C END OF BLADE SURFACE VELOCITY CALCULATIONS
C START CALCULATION OF WEIGHT FLOW VS. DISTANCE FROM HUB
C
260 DO 370 I=1,MX
      IND=1
      DO 270 K=1,KMX
270 AC(K)=DN(I,K)
      GO TO 290
280 WA(I,1)=.5*WA(I,1)
290 DO 300 K=2,KMX
      J=K-1
      HR=R(I,K)-R(I,J)
      HZ=Z(I,K)-Z(I,J)
      WAS=WA(I,J)*(1.+SA(I,J)*HR+SC(I,J)*HZ)+SB(I,J)*HR+SD(I,J)*HZ
      WASS=WA(I,J)+WAS*(SA(I,K)*HR+SC(I,K)*HZ)+SB(I,K)*HR+SD(I,K)*HZ
300 WA(I,K)=(WAS+WASS)/2.
310 DO 340 K=1,KMX
      TIP= 1.-(WA(I,K)**2+2.*W*ALM-(W*R(I,K))**2)/2./CP/TEMP
      IF(TIP.LE.-0) GO TO 280
      TPPIP= 1.- (2.*W*ALM-(W*R(I,K))**2)/2./CP/TEMP
      DENSITY=TIP**EXPON*RH0-(TIP/TPPIP)**EXPON*PLOSS/AR/TPPIP/TEMP
      1 *32.17*SM(I,K)/SM(MXB,L,K)
      PR(I,K)=DENSITY*AR*TIP*TEMP/32.17/144.
      IF(ZS(I).LE.ZH(I)) GO TO 320
      PSI=ATAN((RS(I)-RH(I))/(ZS(I)-ZH(I)))-1.5708

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      GO TO 330
320 PSI=ATAN((ZH(I)-ZS(I))/(RS(I)-RH(I)))
330 WTHRU=WA(I,K)*CBETA(I,K)*COS(PSI-AL(I,K))
      A=XN
      IF(Z(I,K).LT.ZSPLIT) A=SFACT*XN
      C = 6.283186*R(I,K)-A*TT(I,K)
340 AD(K)=DENSTY*WTHRU*C
      CALL INTGRL(AC(I),AD(I),KMX,WTFI(1))
      IF (ABS(WT-WTFI(KMX)).LE.WTOLER) GO TO 350
      CALL CONTIN (WA(I,1),WTFI(KMX),IND,I,WT)
      IF (IND.NE.6) GO TO 290
350 CALL SPLINT (WTFI,AC,KMX,BA,KMX,AB)
      DO 360 K=1,KMX
      DELTA=ABS(AB(K)-DN(I,K))
      DN(I,K)=(1.-CORFAC)*DN(I,K)+CORFAC*AB(K)
360 IF(DELTA.GT.ERROR)ERROR=DELTA
370 CONTINUE
C
C END OF LOOP - WEIGHT FLOW CALCULATION
C CALCULATE STREAMLINE COORDINATES FOR NEXT ITERATION
C
      DO 380 K=2,KMXM1
      DO 380 I=1,MX
      Z(I,K)=DN(I,K)/DN(I,KMX)*(ZS(I)-ZH(I))+ZH(I)
380 R(I,K)=DN(I,K)/DN(I,KMX)*(RS(I)-RH(I))+RH(I)
      IF((ERRJR.GF.ERRJRI).OR.(ERROR.LE.TOLER)) ITER=ITER-1
      IF(ITER.GT.0) GO TO 410
      WRITE (6,1100)
      DO 400 K=1,KMX,NPRT
      WRITE (6,1080) K
      DO 390 I=1,MX
      AB(I)=(Z(I,K)-R(I,K))/ROOT
390 AC(I)=(Z(I,K)+R(I,K))/ROOT
      CALL SPLINE (AB,AC,MX,AD,CURV(1,K))
      DO 400 I=1,MX
      CURV(I,K)=CURV(I,K)/(1.+AD(I) **2)**1.5
      A=DN(I,K)*12.
      B= Z(I,K)*12.
      D= R(I,K)*12.
470 WRITE (6,1110) A,B,D,WA(I,K),PRS(I,K),WTR(I,K),CURV(I,K)
      WRITE (6,1130)
410 A=ERRJR*12.
      WRITE (6,1120) ITNO,A
      ITNO=ITNO+1
      IF (ITER.GE.0) GO TO 150
      WRITE (6,1140)
      K = (KMX+1)/2
      DO 440 I=1,MX
      IF(ZS(I).LE.ZH(I)) GO TO 420
      PSI = ATAN((RS(I)-RH(I))/(ZS(I)-ZH(I)))-1.5708
      GO TO 430
420 PSI = ATAN((ZH(I)-ZS(I))/(RS(I)-RH(I)))
430 AB(I) = (DN(I,2 )-DN(I,1 )) *COS(PSI-AL(I,2))
      AC(I) = (DN(I,K+1)-DN(I,K-1))/2.*COS(PSI-AL(I,K))
      AD(I) = (DN(I,KMX)-DN(I,KMXM1)) *COS(PSI-AL(I,KMX-1))
      A = AB(I)*12.

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      B = AC(I)*12.
      D = AD(I)*12.
440 WRITE (6,1110) A,B,D
      CALL SPLINT (XT,THTA,MHTA,Z(1,1),MX,THH)
      CALL SPLINT (XT,THTA,MHTA,Z(1,2),MX,THH)
      CALL SPLINT (XT,THTA,MHTA,Z(1,K),MX,THM)
      CALL SPLINT (XT,THTA,MHTA,Z(1,KMX),MX,THS)
      PION = 3.1415927/XN
      TPION = PION*SFACF
      WRITE (6,1150)
      DO 450 I=1,4X
        THH1(I) = THH(I)-TPION*TT(I,1)/2./R(I,1)
        THM1(I) = THM(I)-TPION*TT(I,K)/2./R(I,K)
        THS1(I) = THS(I)-TPION*TT(I,KMX)/2./R(I,KMX)
        THHKH(I) = THH(I)-PION*TT(I,1)/2./R(I,1)
        THMKH(I) = THM(I)-PION*TT(I,K)/2./R(I,K)
        THSKH(I) = THS(I)-PION*TT(I,KMX)/2./R(I,KMX)
        THHKP(I) = THH(I)-PION*TT(I,1)/2./R(I,1)
        THMKP(I) = THM(I)-PION*TT(I,K)/2./R(I,K)
        THSKP(I) = THS(I)-PION*TT(I,KMX)/2./R(I,KMX)
        THHKMX(I) = THH(I)-TT(I,1)/2./R(I,1)
        THMKMX(I) = THM(I)-TT(I,K)/2./R(I,K)
        THSKMX(I) = THS(I)-TT(I,KMX)/2./R(I,KMX)
450 WRITE(6,1160) THH1(I),THHKH(I),THHKP(I),THHKMX(I),THM1(I),THMKH(I),
      1,THMKP(I),THMKMX(I),THS1(I),THSKH(I),THSKP(I),THSKMX(I)
      DO 460 J=1,3
        I=1
        K=1
        IF(J.EQ.2) K=(KMX+1)/2
        IF(J.EQ.3) K=KMX
        TIP= 1.-(WA(I,K)**2+2.*W*ALM-(W*R(I,K))**2)/2./CP/TEMP
        DENSTY = TIP**EXPON*RHO
        C = 6.283186*R(I,K)-XN*TT(I,K)*SFACF
        WIDTH = AB
        IF(J.EQ.2) WIDTH = AC
        IF(J.EQ.3) WIDTH = AD
        WM = 8A(2)/DENSTY/C/WIDTH
        WTHETA = ALM/R(I,K)-W*R(I,K)
        BETAI(J) = ATAN(WTHETA/WM)
        AA(J) = BETAI(J)*57.29577
460 CONTINUE
      WRITE (6,1170) AA
      IF(BCDP.EQ.2) CALL RCDUMP(SRW,BETAI(3))
      IF(BCDP.NE.1) RETURN
      CALL RCDUMP (DN(1,1),DN(21,21))
      CALL RCDUMP (WA(1,1),WA(21,21))
      CALL RCDUMP ( Z(1,1), Z(21,21))
      CALL RCDUMP ( R(1,1), R(21,21))
      RETURN
1010 FORMAT (4I5,6F10.4)
1020 FORMAT (8H1RUN NO.I3,10X,25HINPUT DATA CARD LISTING )
1030 FORMAT (7F10.4)
1040 FORMAT (10X24HBCD CARDS FOR DN,WA,Z,R )
1050 FORMAT (36HX STAG. SPEED OF SOUND AT INLET = ,F9.2)
1060 FORMAT (///5X13HITERATION NO.I3)
1070 FORMAT (1H 6X5HALPHA9X5HRC 9X5HSM 9X5HBETA 9X5HTT 9X5H 9

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1X5HSB 9X5HSC 9X5HSD )
1080 FORMAT (2X10HSTREAMLINEI3)
1090 FORMAT (9F14.6)
1100 FORMAT (1HL9X5HON 15X5HZ 15X54R 15X5HWA 15X5HPRESS14X3HW
1TR14X3HRC )
1110 FORMAT (6F19.6,F18.6)
1120 FORMAT (18H ITERATION NO. I3,10X,24HMAX. STREAMLINE CHANGE = ,
1F10.6)
1130 FORMAT (1HJ)
1140 FORMAT (34H1 STREAMLINE SPACING ALONG NORMAL//10X5HHUB 14X5HMEAN
1 14X6HSHROUD)
1150 FORMAT (1H1,56X,17HBLADE COORDINATES/19X,6HHUB ,38X,6HMEAN ,
1 38X,6HSHROUD/3(5X,1H1,10X,4HKHMX, 7X,4HK4P1,7X,3HKMX,3X))
1160 FORMAT (12F11.4)
1170 FORMAT (///1HL,10X,20HINLET ANGLES - HUB,F7.2,8H, MEANF7.2,10H
1, SHROUDF7.2)
END

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$IBFTC SPLBLD DECK
SUBROUTINE SPLBLD(HUB,MEAN,SHROUD,MOB)
COMMON SRW,KMX,WT,XN,GAM,AR,TYPE,BCDP,TEMP,ALM,RHO,TOLER,
1 PLOSS,NPRT,ITER,BETIN,WTOLER,T4H1,T4KH,THHKP,THHKMX,THM1,
2 THMKH,THMKP,THMKMX,THS1,THSKH,T4SKP,THSKMX,Z1,R1,AB,AD,AE,
3 RUNO,MXBL,PION,W,BETAI,OTDM,CURV
DIMENSION Z(21),R(21),DN(21),SM(21),BA(22),AB(22),AC(22),AL(21),
IRC(21),CAL(21),SAL(21),PRS(22),WTF(22),WTHETA(21,22),DNDM(21,22),
2THETA(21,22),WA(21,22),BETA(21,22),SBETA(21,22),CBETA(21,22),
3SA(21,22),SH(21,22),CURV(21,22),DTOM(21,22)
DIMENSION Z1(21,21),R1(21,21), T4H1(21),THHKH(21),THHKP(21),
1 THHKMX(21),THM1(21),THMKH(21),THMKP(21),THMKMX(21),THS1(21),
2 THSKH(21),THSKP(21),THSKMX(21)
DIMENSION THAL(23,4),SMAL(23),AD(22),AE(22),DENSTY(22),BETAI(3)
INTEGER RUNO,TYPE,BCDP,SRW,RR,HUB,SHROUD
LOGICAL PRINT
IF(MDB.EQ.2) GO TO 10
RUNO=0
IF(MDB.EQ.4) GO TO 10
5 READ (5,1010)MX,KMX,MXSP,RR,W,WT,XN,GAM,AR

ITNO = 1
RUNO=RUNO+1
WRITE (6,1020) RUNO
WRITE (6,1010)MX,KMX,MXSP,RR,W,WT,XN,GAM,AR
READ (5,1010)TYPE,BCDP,SRW,MXBL,TEMP,ALM,RHO,TOLER,PLOSS,DNDM1

WRITE(6,1010)TYPE,BCDP,SRW,MXBL,TEMP,ALM,RHO,TOLER,PLOSS,DNDM1
PLOSS=PLOSS*144.
READ (5,1010) NULL,NPRT,ITER,NULL,BETIN,WTOLER,CORFAC

WRITE(6,1010) NULL,NPRT,ITER,NULL,BETIN,WTOLER,CORFAC
BETIN = BETIN/57.29577
KHM1 = KMX/2
KHP1 = KHM1+1
KHP2 = KHM1+2
READ (5,1030) (THETA(I,1),I=1,MX)
WRITE(6,1030) (THETA(I,1),I=1,MX)
READ (5,1030) (THETA(I,KMX),I=1,MX)
WRITE(6,1030) (THETA(I,KMX),I=1,MX)
READ (5,1030) (THETA(I,KHM1),I=1,MXSP)
WRITE(6,1030) (THETA(I,KHM1),I=1,MXSP)
READ (5,1030) (THETA(I,KHP1),I=1,MXSP)
WRITE(6,1030) (THETA(I,KHP1),I=1,MXSP)
IF(RR.EQ.1) GO TO 90
READ (5,1030) (Z(I),I=1,MX)
WRITE(6,1030) (Z(I),I=1,MX)
READ (5,1030) (R(I),I=1,MX)
WRITE(6,1030) (R(I),I=1,MX)
READ (5,1030) (DN(I),I=1,MX)
WRITE(6,1030) (DN(I),I=1,MX)
DO 9 I=1,MX
Z(I)=Z(I)/12.
R(I)=R(I)/2.
9 DN(I)=DN(I)/12.
GO TO 21

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10 IF(MQB.EQ.4) CALL BCREAD(SRW,BETA(3))
    MX = MX+1
    MQB = 2
    WT = WT/FLOAT(KMXF-1)
    WTOLER = WTOLER/FLOAT(KMXF-1)
    MXBL = MXBL+1
11 READ (5,1010)KMX,MXSP,RR ,TYPE,CORFAC,TOLER,THIKMX

    RUNO = RUNO+1
    IF((HUB.EQ.0).AND.(MEAN.EQ.0).AND.(SHROUD.EQ.0)) RETURN
    WRITE (6,1020) RUNO
    WRITE(6,1010)KMX,MXSP,RR ,TYPE,CORFAC,TOLER,THIKMX
    ITNO = 1
    ITER = ITER1
    KHM1 = KMX/2
    KHP1 = KHM1+1
    KHP2 = KHM1+2
    READ (5,1030) Z(1),R(1),DN(1)
    WRITE(6,1030) Z(1),R(1),DN(1)
    Z(1) = Z(1)/12.
    R(1) = R(1)/12.
    DN(1) = DN(1)/12.
    DWD1 = W+ALM/R(1)**2
    IF(HUB.EQ.0) GO TO 14
    HUB = 0
    DO 12 I=2,MX
        J = I-1
        THETA(I,1) = THH1(J)
        THETA(I,KMX) = THHKMX(J)
        Z(I) = Z1(J,1)
        R(I) = R1(J,1)
12 DN(I) = AB(J)
        DO 13 I=2,MXSP
            J = I-1
            THETA(I,KHM1) = THHKH(J)
13 THETA(I,KHP1) = THHKP(J)
            BETIN = BETA(1)
            GO TO 20
14 IF(MEAN.EQ.0) GO TO 17
            MEAN = 0
            K = (KMXF+1)/2
            DO 15 I=2,MX
                J = I-1
                THETA(I,1) = THM1(J)
                THETA(I,KMX) = THMKMX(J)
                Z(I) = Z1(J,K)
                R(I) = R1(J,K)
15 DN(I) = AD(J)
                DO 16 I=2,MXSP
                    J = I-1
                    THETA(I,KHM1) = THMKH(J)
16 THETA(I,KHP1) = THMKP(J)
                    BETIN = BETA(2)
                    GO TO 20
17 IF(SHROUD.EQ.0) RETURN
    SHROUD = 0

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DO 18 I=2,MX
  J = I-1
  THETA(I,1) = THS1(J)
  THETA(I,KMX) = THSKMX(J)
  Z(I) = Z1(J,KMXF)
  R(I) = R1(J,KMXF)
18 DN(I) = AE(J)
DO 19 I=2,MXSP
  J = I-1
  THETA(I,KHMX) = THSKH(J)
19 THETA(I,KHP1) = THSKP(J)
  BETIN = BETAI(3)
20 TANBTO = SIN(BETIN)/COS(BETIN)
  TANBT = TANBTO*(ALM-W*R(1)**2)/(ALM-W*R(2)**2)*DN(1)/DN(2)
  DTDM1 = TANBT/R(1)
  DELTHT = TANBTO/(W*R(2)**2-ALM)*(ALM*ALOG(R(1)/R(2))+W/2.*(R(2)**2
1  -R(1)**2))*(DN(1)+DN(2))/2./DN(2)
  THETA(I,KMX) = THIKMX*DELTHT
  THETA(I,KHP1) = THETA(I,KMX)-PI*DN
  THETA(I,KHMX) = THETA(I,KHP1)
  THETA(I,1) = THETA(I,KHMX)-PI*DN
  NULL = 0
  WRITE(6,1010)MX,KMX,MXSP,RR,W,WT,XN,GAM,AR
  PLOSS1 = PLTSS/144.
  WRITE(6,1010)TYPE,BCDP,SRW,MXBL,TEMP,ALM,RHO,TOLER,PLOSS1,DWDM1
  BETIN1 = BETIN*57.29577
  WRITE(6,1010) NULL,NPRI,ITER,NULL,BETIN1,WTOLER,CORFAC
  WRITE(6,1030) (THETA(I,1),I=1,MX)
  WRITE(6,1030) (THETA(I,KMX),I=1,MX)
  WRITE(6,1030) (THETA(I,KHMX),I=1,MXSP)
  WRITE(6,1030) (THETA(I,KHP1),I=1,MXSP)
  WRITE(6,1030) (Z(I),I=1,MX)
  WRITE(6,1030) (R(I),I=1,MX)
  WRITE(6,1030) (DN(I),I=1,MX)
21 IF(TYPE.EQ.1) GO TO 30
  WA(1,1)=WT/RHO/DN(1)/R(1)/XN/(THETA(1,KMX)-THETA(1,1))
  DO 23 I=1,MX
    IF((I.EQ.1).OR.(I.GT.MXSP)) THETA(I,KHMX) = (THETA(I,KMX)+THETA
1  (I,1))/2.
    IF((I.EQ.1).OR.(I.GT.MXSP)) THETA(I,KHP1) = THETA(I,KHMX)
    DO 22 K=1,KHMX
      THETA(I,K) = FLOAT(K-1)/FLOAT(KHMX-1)*(THETA(I,KHMX)-THETA(I,1))
1  + THETA(I,1)
22 WA(I,K)=WA(1,1)
    DO 23 K=KHP1,KMX
      THETA(I,K) = FLOAT(K-KHP1)/FLOAT(KMX-KHP1)*(THETA(I,KMX)-
1  THETA(I,KHP1))+THETA(I,KHP1)
23 WA(I,K) = WA(1,1)
    READ(5,1010) NEXT
    WRITE(6,1010) NEXT
24 IF(NEXT.EQ.0) GO TO 25
  READ(5,1021) I1,K1,THETA(I1,K1),I2,K2,THETA(I2,K2),I3,K3,THETA(I3,
1  K3),I4,K4,THETA(I4,K4),I5,K5,THETA(I5,K5),NEXT

  WRITE(6,1021)I1,K1,THETA(I1,K1),I2,K2,THETA(I2,K2),I3,K3,THETA(I3,
1  K3),I4,K4,THETA(I4,K4),I5,K5,THETA(I5,K5),NEXT

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```

      GO TO 24
C
C      END OF INPUT STATEMENTS
C
25 WRITE (6,1023)
   DO 26 K=1,KMX
26 WRITE (6,1030) (THETA(I,K),I=1,MX)

      GO TO 45
30 CALL BCREAD (THETA(1,1),THETA(21,21))
   CALL BCREAD (WA (1,1), WA(21,21))
   WRITE (6,1040)
45 CONTINUE
   CP=AR*GAM/(GAM-1.)
   CI = SQRT(GAM*AR*TEMP)
   WRITE (6,1050) CI
   KMXM1 = KMX-1
   MXP2 = MX+2
   MXSP1 = MXSP+1
   WTH = WT/2.
   EXPON = 1./(GAM-1.)
   DTOM1=SIN(BETIN)/COS(BETIN)/R(1)
   IF(MJB.EQ.2) DTD1 = TANBT/R(1)
   ROOT=SQRT(2.)
   SM(1) = -SQRT((Z(2)-Z(1))**2+(R(2)-R(1))**2)
   DO 60 K=1,KMXM1
60 BA(K)=FLOAT(K-1)*WT/FLOAT(KMXM1-1)

C
C      CALCULATE ALPHA AND SM
C
      DO 70 I=1,MX
      AB(I)=(Z(I)-R(I))/ROOT
70 AC(I)=(Z(I)+R(I))/ROOT
   CALL SPLINE(AB,AC,MX,4L,RC)
   DO 80 I=1,MX
   AL(I)=ATAN(AL(I))- .785398
   CAL(I)=COS(AL(I))
80 SAL(I)=SIN(AL(I))
   DO 85 I=2,MX
   J=I-1
85 SM(I)=SM(J)+SQRT((Z(I)-Z(J))**2+(R(I)-R(J))**2)
   SMAL(3) = SM(2)+.1*(SM(3)-SM(2))
   SMAL(4) = SM(2)+.5*(SM(3)-SM(2))
90 ERROR = 1000.

C
C      CALCULATE BETA ON BLADE SURFACES
C      BEGINNING OF LOOP FOR ITERATIONS
C
91 DO 97 J=1,4
   K = 1
   IF(J.EQ.2) K=KMX
   IF(J.EQ.3) K=KHP1
   IF(J.EQ.4) K=KMX
   DO 93 I=1,2
   THAL(I,J) = THETA(I,K)
93 SMAL(I) = SM(I)

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DO 94 I=3,4
94 THAL(I,J) = THETA(2,K)+(SMAL(I)-SM(2))/(SM(3)-SM(2))*(THETA(3,K)
1 -THETA(2,K))
DO 95 I=5,MXP2
IM2 = I-2
THAL(I,J) = THETA(IM2,K)
95 SMAL(I) = S4(IM2)
CALL SPLIN2(SMAL,THAL(1,J),DTDM1,MXP2,DTDM(1,K),AB)
DO 96 I=1,2
CURV(I,K) = AB(I)/12./((1.+(DTDM(I,K)/12.))**2)**1.5
BETA(I,K) = ATAN(R(I)*DTDM(I,K))
SBETA(I,K) = SIN(BETA(I,K))
CBETA(I,K) = COS(BETA(I,K))
96 CONTINUE
DO 97 I=3,MX
IP2 = I+2
CURV(I,K) = AB(IP2)/12./((1.+(DTDM(IP2,K)/12.))**2)**1.5
BETA(I,K) = ATAN(R(I)*DTDM(IP2,K))
SBETA(I,K) = SIN(BETA(I,K))
CBETA(I,K) = COS(BETA(I,K))
97 CONTINUE
PRINT = (ITER.LE.0).OR.(ITNO.LE.NC)
IF (PRINT) WRITE(6,1060)ITNO
ERROR1=ERROR
ERROR=0.
C
C START CALCULATION OF PARAMETERS
C
LAST = KMX-1
DO 100 K=2, LAST
CALL SPLIN2(SM,THETA(1,K),DTDM1,MX,DTDM(1,K),AB)
DO 100 I=1,MX
CURV(I,K) = AB(I)/12./((1.+(DTDM(I,K)/12.))**2)**1.5
BETA(I,K)=ATAN(R(I)*DTDM(I,K))
SBETA(I,K)=SIN(BETA(I,K))
CBETA(I,K)=COS(BETA(I,K))
100 CONTINUE
DO 101 K=KHP2,KMXM1
CALL SPLIN2(SM,THETA(1,K),DTDM1,MX,DTDM(1,K),AB)
DO 101 I=1,MX
CURV(I,K) = AB(I)/12./((1.+(DTDM(I,K)/12.))**2)**1.5
BETA(I,K)=ATAN(R(I)*DTDM(I,K))
SBETA(I,K)=SIN(BETA(I,K))
CBETA(I,K)=COS(BETA(I,K))
101 CONTINUE
DO 110 K=1,KMX
DO 105 I=1,MX
WTHETA(I,K)=WA(I,K)*SBETA(I,K)
105 CONTINUE
CALL SPLIN2(SM,WTHETA(1,K),DWDM1,MX,DWDM(1,K),AC)
DO 110 I=1,MX
SA(I,K) = SAL(I)*SBETA(I,K)*CBETA(I,K)
SB(I,K) = CBETA(I,K)*R(I)*(2.*W*SAL(I)+DWDM(I,K))
110 CONTINUE
C
C END OF PARAMETER CALCULATION

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C CALCULATE VELOCITY DISTRIBUTION,CHECK CONTINUITY
C

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DO 200 I=1,MX
IND=1
IND2 = 1
GO TO 130
120 WA(I,1)=.5*WA(I,1)
GO TO 130
125 WA(I,1)=2.*WA(I,1)
GO TO 130
126 WA(I,KHP1) = .5*WA(I,KHP1)
GO TO 130
127 WA(I,KHP1) = 2.*WA(I,KHP1)
130 IF((I.GT.1).AND.(I.LE.MXSP)) GO TO 142
DO 140 K=2,KMX
J = K-1
IF(K.EQ.KHP1) WA(I,K) = WA(I,J)
IF(K.EQ.KHP1) GO TO 140
HT=THETA(I,K)-THETA(I,J)
WAS = WA(I,J)+(WA(I,J)*SA(I,J)+SB(I,J))*HT
WASS = WA(I,J)+(WAS*SA(I,K)+SB(I,K))*HT
WA(I,K) = (WAS+WASS)/2.
140 CONTINUE
GO TO 148
142 DO 144 K=2,KHMX
J=K-1
HT=THETA(I,K)-THETA(I,J)
WAS = WA(I,J)+(WA(I,J)*SA(I,J)+SB(I,J))*HT
WASS = WA(I,J)+(WAS*SA(I,K)+SB(I,K))*HT
WA(I,K) = (WAS+WASS)/2.
144 CONTINUE
DO 146 K=KHP2,KMX
J = K-1
HT=THETA(I,K)-THETA(I,J)
WAS = WA(I,J)+(WA(I,J)*SA(I,J)+SB(I,J))*HT
WASS = WA(I,J)+(WAS*SA(I,K)+SB(I,K))*HT
WA(I,K) = (WAS+WASS)/2.
146 CONTINUE
148 CONTINUE
DO 150 K=1,KMX
TIP= 1.-(WA(I,K)**2+2.*W*ALM-(W*R(I) )**2)/2./CP/TEMP
IF((TIP.LT..0).AND.(I.GT.1).AND.(I.LE.MXSP).AND.(K.GE.KHP1))
GO TO 126
IF(TIP.LT..0) GO TO 120
TPP1P= 1.-(2.*W*ALM-(W*R(I) )**2)/2./CP/TEMP
DENSTY(K) = TIP**EXPON*RHO-(TIP/TPP1P)**EXPON*PLOSS/AR/TPP1P/TEMP
1 *32.17*SM(I) /SM(MXRL)
PRS(K) = DENSTY(K)*AR*TIP*TEMP/32.17/144.
WM=WA(I,K)*CBETA(I,K)
AB(K) = DENSTY(K)*WM*DN(I)*R(I)*XN
150 AC(K)=THETA(I,K)
CALL INTGRL(AC,AB,KHMX,WTFL)
IF(WTFL(KHMX).LE..0) GO TO 125
CALL INTGRL(AC(KHP1),AB(KHP1),KHMX,WTFL(KHP1))
IF((WTFL(KMX).LE..0).AND.(I.GT.1).AND.(I.LE.MXSP)) GO TO 127
IF(WTFL(KMX).LE..0) GO TO 125

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      IF((I.GT.1).AND.(I.LE.MXSP)) GO TO 155
      DO 152 K=KHP1,KMXM1
      AC(K) = AC(K+1)
152  WTFL(K) = WTFL(KHMX)+WTFL(K+1)
      IF(ABS(WT-WTFL(KMXM1)).LE.WTOLFR) GO TO 160
      CALL CONTIN(WA(I,1),WTFL(KMXM1),IND,I,WT)
      IF(IND.NE.6) GO TO 130
      GO TO 160
155  IF (ABS(WTH-WTFL(KHMX)).LE.WTOLFR) IND=6
      IF (ABS(WTH-WTFL(KMX)).LE.WTOLFR) IND2=6
      IF(IND.NE.6) CALL CONTIN(WA(I,1),WTFL(KHMX),IND,I,WTH)
      IF(IND2.NE.6) CALL CONTAL(WA(I,KHP1),WTFL(KMX),IND2,I,WTH)
      IF((IND.NE.6).OR.(IND2.NE.6)) GO TO 130
160  IF((I.GT.1).AND.(I.LE.MXSP)) GO TO 165
      CALL SPLINT (WTFL,AC,KMXM1,BA,KMXM1,AB)
      GO TO 166
165  CALL SPLINT(WTFL,AC,KHMX,BA,KHMX,AB)
      CALL SPLINT(WTFL(KHP1),AC(KHP1),KHMX,BA(1),KHMX,AB(KHP1))
166  CONTINUE
      IF((I.GT.1).AND.(I.LE.MXSP)) GO TO 168
      DO 167 K=1,KHMX
      KA = KMX-K+1
      J = KA-1
167  AB(KA) = AB(J)
168  DO 170 K=1,KMX
      DELTA=ABS(AB(K)-THETA(I,K))
170  IF(DELTA.GT.ERROR)ERROR=DELTA
      IF(.NOT.PRINT) GO TO 178
      A=SN(I)*12.
      C=AL(I)*57.29577
      D = R(I)*12.
      E = Z(I)*12.
      F = DN(I)*12.
      WRITE (6,1080) I,A,C,D,E,F
      WRITE (6,1070)
      DO 175 K=1,KHMX,NPRT
      B=BETA(I,K)*57.29577
      C = WTHETA(I,K)+W*R(I)
      WM = WA(I,K)*CBETA(I,K)
      V = SQRT(C**2+WM**2)
      OWDT = WA(I,K)*SA(I,K)+SB(I,K)
      WRITE (6,1090) THETA(I,K),CURV(I,K),B,WA(I,K),WTHETA(I,K),C,WM,V,
1  PRS(K),DENSTY(K),DTDM(I,K),DWDM(I,K),SA(I,K),SB(I,K),DWDT
175  CONTINUE
      WRITE (6,1095)
      DO 176 K=KHP1,KMX,NPRT
      B=BETA(I,K)*57.29577
      C = WTHETA(I,K)+W*R(I)
      WM = WA(I,K)*CBETA(I,K)
      V = SQRT(C**2+WM**2)
      DWDT = WA(I,K)*SA(I,K)+SB(I,K)
      WRITE (6,1090) THETA(I,K),CURV(I,K),B,WA(I,K),WTHETA(I,K),C,WM,V,
1  PRS(K),DENSTY(K),DTDM(I,K),DWDM(I,K),SA(I,K),SB(I,K),DWDT
176  CONTINUE
      WRITE (6,1095)
178  DO 180 K=2,KMXM1

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180 THETA(I,K)=(1.-CORFAC)*THETA(I,K)+CORFAC*AB(K)
200 CONTINUE
C
C   END OF VELOCITY CALCULATIONS.
C
WRITE (6,1120) ITNO,ERROR
IF(ITER.LE.0) GO TO 230
IF((ERROR.GE.ERROR1).OR.(ERROR.LE.TOLER)) ITER=ITER-1
ITNO=ITNO+1
GO TO 91
230 IF(BCDP.NE.1) GO TO 240
CALL BCDUMP (THETA(1,1),THETA(21,21))
CALL BCDUMP ( WA(1,1), WA(21,21))
240 IF(MJR.EQ.2) GO TO 11
GO TO 5
1010 FORMAT (4I5,6F10.4)
1020 FORMAT (8H1RUN NO.13,10X,25HINPUT DATA CARD LISTING )
1021 FORMAT (5(2I2,F8.5),11)
1023 FORMAT (32H THETA-CALCULATED AND/OR INPUT )
1030 FORMAT (7F10.4)
1040 FORMAT (10X25HBCD CARDS FOR THETA,WA )
1050 FORMAT (36HK STAG. SPEED OF SOUND AT INLET = ,F9.2,/)
1060 FORMAT (///5X13HITERATION NO.13)
1070 FORMAT (132HK THETA T-CURV BETA WA WTHETA VTHETA
1WH V PRS DENSITY DTDM DWDM SA SB D4
2DT )
1080 FORMAT (2X16HQUASI-ORTHOGONAL13,6X,4HSM =,F7.4,9H ALPHA =, F7.2,
1 5H R =,F7.4,5H Z =,F7.4,6H DN =,F7.4)
1090 FORMAT (1X,F9.4,2F8.2,5F8.1,F7.2,F9.5,F7.2,F8.0,F8.4,2F9.1)
1095 FORMAT (1H )
1120 FORMAT (18HJ ITERATION NO. 13,10X,24HMAX. STREAMLINE CHANGE = ,
1F10.6)
1200 FORMAT(///10X7HNORMAL 14)
1210 FORMAT(7F18.6)
END

```

```

$IBFTC SPLINT DECK
SUBROUTINE SPLINT (X,Y,N,Z,MAX,YINT)
  DIMENSION X(50),Y(50),S(50),A(50),B(50),C(50),F(50),W(50),SB(50),
  IG(50),EM(50),Z(50),YINT(50)
  COMMON Q
  INTEGER Q
  DO 10 I=2,N
    S(I)=X(I)-X(I-1)
  NO=N-1
  DO 20 I=2,N
    A(I)=S(I)/6.0
    B(I)=(S(I)+S(I+1))/3.0
    C(I)=S(I+1)/6.0
  20 F(I)=(Y(I+1)-Y(I))/S(I+1)-(Y(I)-Y(I-1))/S(I)
    A(N)=-.5
    B(1)=1.0
    B(N)=1.0
    C(1)=-.5
    F(1)=0.0
    F(N)=0.0
    W(1)=B(1)
    SB(1)=C(1)/W(1)
    G(1)=0.0
    DO 30 I=2,N
      W(I)=B(I)-A(I)*SB(I-1)
      SB(I)=C(I)/W(I)
  30 G(I)=(F(I)-A(I)*G(I-1))/W(I)
    EM(N)=G(N)
    DO 40 I=2,N
      K=N+1-I
  40 EM(K)=G(K)-SB(K)*EM(K+1)
    DO 90 I=1,MAX
      K=2
      IF(Z(I)-X(1)) 60,50,70
  50 YINT(I)=Y(1)
      GO TO 90
  60 IF(Z(I).LT.(1.1*X(1)-.1*X(2)))WRITE (6,1000)Z(I)
      GO TO 85
  1000 FORMAT (17H OUT OF RANGE Z =F10.6)
  65 IF(Z(I).GT.(1.1*X(N)-.1*X(N-1))) WRITE (6,1000)Z(I)
      K=N
      GO TO 85
  70 IF(Z(I)-X(K)) 85,75,80
  75 YINT(I)=Y(K)
      GO TO 90
  80 K=K+1
      IF(K=N) 70,70,65
  85 YINT(I) = EM(K-1)*(X(K)-Z(I))*3/6./S(K)+EM(K)*(Z(I)-X(K-1))*3/6.
      1/S(K)+(Y(K)-Y(K-1))/S(K)-EM(K)*S(K)/6.)*(Z(I)-X(K-1))+(Y(K-1)-Y(K-2))/S(K)-EM(K-1)
      2*S(K)/6.)*(X(K)-Z(I))
  90 CONTINUE
    MXA = MAX0(N,MAX)
    IF(Q.EQ.16) WRITE(6,1010) N,MAX,(X(I),Y(I),Z(I),YINT(I),I=1,MXA)
  1010 FORMAT (2X21HNO. OF POINTS GIVEN =,I3,30H, NO. OF INTERPOLATED POI
    INTS =,I3,/10X5HX 15X5HY 12X11HX-INTERPOL.9X11HY-INTERPOL./14
    2E20.8))
  100 RETURN
  END

```

```

$IBFTC SPLDER  DECK
SUBROUTINE SPLDER(X,Y,N,Z,MAX,DYDX)
DIMENSION X(50),Y(50),S(50),A(50),B(50),C(50),F(50),W(50),SB(50),
IG(50),FM(50),Z(50),DYDX(50)
DO 10 I=2,N
10 S(I)=X(I)-X(I-1)
ND=N-1
DO 20 I=2,ND
A(I)=S(I)/6.0
B(I)=(S(I)+S(I+1))/3.0
C(I)=S(I+1)/6.0
20 F(I)=(Y(I+1)-Y(I))/S(I+1)-(Y(I)-Y(I-1))/S(I)
A(N)=-.5
B(1)=1.0
B(N)=1.0
C(1)=-.5
F(1)=0.0
F(N)=0.0
W(1)=B(1)
SB(1)=C(1)/W(1)
G(1)=0.0
DO 30 I=2,N
W(I)=B(I)-A(I)*SB(I-1)
SB(I)=C(I)/W(I)
30 G(I)=(F(I)-A(I)*G(I-1))/W(I)
EM(N)=G(N)
DO 40 I=2,N
K=N+1-I
40 EM(K)=G(K)-SB(K)*EM(K+1)
DO 90 I=1,MAX
K=2
IF(Z(I)-X(1)) 60,70,70
60 WRITE (6,1000)Z(I)
1000 FORMAT (17H OUT OF BLADE Z =F10.6)
GO TO 85
65 WRITE (6,1000)Z(I)
K=N
GO TO 85
70 IF(Z(I)-X(K)) 85,85,80
80 K=K+1
IF(K=N) 70,70,65
85 DYDX(I)=-EM(K-1)*(X(K)-Z(I))**2/2.0/S(K)+EM(K)*(X(K-1)-Z(I))**2/2.
10/S(K)+(Y(K)-Y(K-1))/S(K)-(EM(K)-EM(K-1))*S(K)/6.0
90 CONTINUE
100 RETURN
END

```



```

$IBFTC CONTIN DECK
      SUBROUTINE CONTIN (WA,WTFL,IND,I,WT)
      DIMENSION SPEED(3),WEIGHT(3)
135 GO TO (140,150,210,270,370),IND
140 SPEED(1) = WA
      WEIGHT(1) = WTFL
      DELTA = WT/WTFL*WA-WA
      IF (ABS(DELTA).GT.100.) DELTA = SIGN(100.,DELTA)
      WA = DELTA+WA
      IND = 2
      RETURN
150 IF ((WTFL-WEIGHT(1))/(WA-SPEED(1))) 180,180,160
160 SPEED(2) = WA
      DELTA = (WT-WTFL)/(WTFL-WEIGHT(1))*(WA-SPEED(1))
      IF (ABS(DELTA).GT.100.) DELTA = SIGN(100.,DELTA)
      WA = DELTA+WA
166 SPEED(1) = SPEED(2)
      WEIGHT(1) = WTFL
      RETURN
170 WRITE (6,1000) I,WTFL
      IND = 6
      RETURN
180 IND = 3
      IF (WTFL.GE.WT) GO TO 140
      IF (SPEED(1)-WA) 190,200,200
190 SPEED(2) = SPEED(1)
      SPEED(1) = 2.0*SPEED(1)-WA
      SPEED(3) = WA
      WEIGHT(2) = WEIGHT(1)
      WEIGHT(3) = WTFL
      WA = SPEED(1)
      RETURN
200 SPEED(2) = WA
      SPEED(3) = SPEED(1)
      SPEED(1) = 2.0*WA-SPEED(1)
      WEIGHT(2) = WTFL
      WEIGHT(3) = WEIGHT(1)
      WA = SPEED(1)
      RETURN
210 WEIGHT(1) = WTFL
      IF (WTFL.GE.WT) GO TO 140
      IF (WEIGHT(1)-WEIGHT(2)) 230,380,220
220 WEIGHT(3) = WEIGHT(2)
      WEIGHT(2) = WEIGHT(1)
      SPEED(3) = SPEED(2)
      SPEED(2) = SPEED(1)
      SPEED(1) = 2.0*SPEED(2)-SPEED(3)
      WA = SPEED(1)
      RETURN
230 IF (SPEED(3)-SPEED(1)-10.0) 170,170,240
240 IND = 4
245 IF (WEIGHT(3)-WEIGHT(1)) 260,260,250
250 WA = (SPEED(1)+SPEED(2))/2.0
      RETURN
260 WA = (SPEED(3)+SPEED(2))/2.0
      RETURN

```

```

270 IF (SPEED(3)-SPEED(1)-10.0) 170,170,280
280 IF (WTFL-WEIGHT(2)) 320,350,290
290 IF (WA-SPEED(2)) 310,300,300
300 SPEED(1) = SPEED(2)
    SPEED(2) = WA
    WEIGHT(1) = WEIGHT(2)
    WEIGHT(2) = WTFL
    GO TO 245
310 SPEED(3) = SPEED(2)
    SPEED(2) = WA
    WEIGHT(3) = WEIGHT(2)
    WEIGHT(2) = WTFL
    GO TO 245
320 IF (WA-SPEED(2)) 340,330,330
330 WEIGHT(3) = WTFL
    SPEED(3) = WA
    GO TO 245
340 WEIGHT(1) = WTFL
    SPEED(1) = WA
    GO TO 245
350 IND = 5
    IF (WA-SPEED(2)) 380,360,360
360 SPEED(1) = SPEED(2)
    WEIGHT(1) = WEIGHT(2)
    SPEED(2) = (SPEED(1)+SPEED(3))/2.0
    WA = SPEED(2)
    RETURN
370 IND = 4
    WEIGHT(2) = WTFL
    WA = (SPEED(1)+SPEED(2))/2.0
    RETURN
380 IND = 5
390 WEIGHT(3) = WEIGHT(2)
    SPEED(3) = SPEED(2)
    SPEED(2) = (SPEED(1)+SPEED(3))/2.0
    WA = SPEED(2)
    RETURN
1000 FORMAT (/12H FIXED LINE 12,12H, MAX WT = F10.6)
END

```

```

$IBFTC SPLIN2 DECK
SUBROUTINE SPLIN2(X,Y,Y1P,N,SLOPE,EM)
  DIMENSION X(50),Y(50),S(50),A(50),B(50),C(50),F(50),W(50),SB(50),
  IG(50),EM(50),SLOPE(50)
  COMMON Q
  INTEGER Q
  DO 10 I=2,N
10  S(I)=X(I)-X(I-1)
    NO=N-1
    DO 20 I=2,NO
20  A(I)=S(I)/6.
    B(I)=(S(I)+S(I+1))/3.
    C(I)=S(I+1)/6.
    F(I)=(Y(I+1)-Y(I))/S(I+1)-(Y(I)-Y(I-1))/S(I)
    A(N)=-.5
    B(1)=S(2)/3.
    B(N)=1.
    C(1)=S(2)/6.
    F(1)=(Y(2)-Y(1))/S(2)-Y1P
    F(N)=0.
    W(1)=B(1)
    SB(1)=C(1)/W(1)
    G(1)=F(1)/W(1)
    DO 30 I=2,N
30  W(I)=B(I)-A(I)*SB(I-1)
    SB(I)=C(I)/W(I)
    G(I)=(F(I)-A(I)*G(I-1))/W(I)
    FM(N)=G(N)
    DO 40 I=2,N
40  EM(K)=G(K)-SB(K)*EM(K+1)
    SLOPE(1)=-S(2)/6.*(2.*EM(1)+EM(2))+(Y(2)-Y(1))/S(2)
    DO 50 I=2,N
50  SLOPE(I)=S(I)/6.*(2.*EM(I)+EM(I-1))+(Y(I)-Y(I-1))/S(I)
    IF (2.EQ.14) WRITE (6,100) N,(X(I),Y(I),SLOPE(I),EM(I),I=1,N)

100 FORMAT (2X15HNO. OF POINTS =I3/10X5HX      15X5HY      15X5HSLOPE15X5H
1EM      /14F20.8))
  RETURN
  END

```

\$IBFTC SP_LINE DECK
 SUBROUTINE SPLINE (X,Y,N,SLOPE,FM)
 DIMENSION X(50),Y(50),S(50),A(50),B(50),C(50),F(50),W(50),SB(50),
 IG(50),FM(50),SLOPE(50)
 COMMON Q
 INTEGER Q
 DO 10 I=2,N
 10 S(I)=X(I)-X(I-1)
 NO=N-1
 DO 20 I=2,NO
 A(I)=S(I)/6.
 B(I)=(S(I)+S(I+1))/3.
 C(I)=S(I+1)/6.
 20 F(I)=(Y(I+1)-Y(I))/S(I+1)-(Y(I)-Y(I-1))/S(I)
 A(N)=-.5
 B(N)=1.
 C(N)=1.
 F(N)=-.5
 F(1)=0.
 F(N)=0.
 G(1)=B(1)
 SB(1)=C(1)/W(1)
 G(1)=0.
 DO 30 I=2,N
 W(I)=B(I)-A(I)*SB(I-1)
 SB(I)=C(I)/W(I)
 30 G(I)=(F(I)-A(I)*G(I-1))/W(I)
 EM(N)=G(N)
 DO 40 I=2,N
 K=N+1-I
 40 E(I)=G(K)-SB(K)*EM(K+1)
 SLOPE(1)=-S(2)/6.*(2.*EM(1)+EM(2))+(Y(2)-Y(1))/S(2)
 DO 50 I=2,N
 50 SLOPE(I)=S(I)/6.*(2.*EM(I)+EM(I-1))+(Y(I)-Y(I-1))/S(I)
 IF (Q.EQ.13) WRITE (6,100) N,(X(I),Y(I),SLOPE(I),FM(I),I=1,N)
 100 FORMAT (2X15HNO. OF POINTS =I3/10X5HX 15X5HY 15X5HSLOPE15X5H
 1EM //4F20.8)
 RETURN
 END

```

$IBFTC INTGRL DECK
      SUBROUTINE INTGRL (X,Y,N,SUM)
      DIMENSION X(50),Y(50),S(50),A(50),B(50),C(50),F(50),W(50),SB(50),
      G(50),EM(50),SUM(50)
      COMMON SRW
      INTEGER SRW
      DO 10 I=2,N
10  S(I)=X(I)-X(I-1)
      NO=N-1
      DO 20 I=2,N
20  A(I)=S(I)/6.0
      B(I)=(S(I)+S(I+1))/3.0
      C(I)=S(I+1)/6.0
20  F(I)=(Y(I+1)-Y(I))/S(I+1)-(Y(I)-Y(I-1))/S(I)
      A(N)=-.5
      B(N)=1.0
      C(N)=1.0
      F(N)=-.5
      F(1)=0.0
      F(N)=0.0
      W(1)=6(1)
      SB(1)=C(1)/W(1)
      G(1)=0.0
      DO 30 I=2,N
30  W(I)=B(I)-A(I)*SB(I-1)
      SB(I)=C(I)/W(I)
30  G(I)=(F(I)-A(I)*G(I-1))/W(I)
      EM(N)=G(N)
      DO 40 I=2,N
40  EM(I)=G(I)-SB(I)*EM(I+1)
      SUM(1)=0.0
      DO 50 K=2,N
50  SUM(K)=SUM(K-1)+S(K)*(Y(K)+Y(K-1))/2.0-S(K)**3*(EM(K)+EM(K-1))/2
      14.0
      IF(SRW.EQ.17) WRITE(6,1000) N,(X(I),Y(I),SUM(I),EM(I),I=1,N)

      RETURN
1000 FORMAT (17H NO. OF POINTS =I3/10X5HX      15X5HY      15X5HSUM
1       13X10H2ND DERIV./(4F20.8))
      END

```

```

$IBFTC LININT DECK
SUBROUTINE LININT(X1,Y1,X,Y,TN,MX,MY,F)
COMMON K
DIMENSION X(MX),Y(MY),TN(MX,MY)
DO 10 J3=1,MX
10 IF(X1.LE.X(J3))GO TO 20
J3=MX
20 DO 30 J4=1,MY
30 IF(Y1.LE.Y(J4))GO TO 40
J4=MY
40 J1=J3-1
J2=J4-1
EPS1=(X1-X(J1))/(X(J3)-X(J1))
EPS2=(Y1-Y(J2))/(Y(J4)-Y(J2))
EPS3=1.-EPS1
EPS4=1.-EPS2
F=TN(J1,J2)*EPS3*EPS4+TN(J3,J2)*EPS1*EPS4+TN(J1,J4)*EPS2*EPS3+
1TN(J3,J4)*EPS1*EPS2
IF(X.EQ.14) WRITE(6,1)X1,Y1,F,J1,J2,EPS1,EPS2
1 FORMAT (8H LININT3F10.5,2I3,2F10.5)
K=0
RETURN
END

```

```

$IBFTC CONTAL DECK
      SUBROUTINE CONTAL (WA,WTFL,IND,I,WT)
      DIMENSION SPEED(3),WEIGHT(3)
135 GO TO (140,150,210,270,370),IND
140 SPEED(1) = WA
      WEIGHT(1) = WTFL
      WA = WT/WTFL*WA
      IND = 2
      RETURN
150 IF ((WTFL-WEIGHT(1))/(WA-SPEED(1))) 180,180,160
160 SPEED(2) = WA
      WA = (WT-WTFL)/(WTFL-WEIGHT(1))
      1 *(WA-SPEED(1))+WA
      IF (ABS(WA-SPEED(2))-100.0) 166,166,161
161 IF(WA-SPEED(2))163,163,162
162 WA = SPEED(2)+100.0
      GO TO 166
163 WA = SPEED(2)-100.0
166 SPEED(1) = SPEED(2)
      WEIGHT(1) = WTFL
      RETURN
170 WRITE (6,1000) I,WTFL
      IND = 6
      RETURN
180 IND = 3
      IF (WTFL.GE.WT) GO TO 140
      IF (SPEED(1)-WA) 190,200,200
190 SPEED(2) = SPEED(1)
      SPEED(1) = 2.0*SPEED(1)-WA
      SPEED(3) = WA
      WEIGHT(2) = WEIGHT(1)
      WEIGHT(3) = WTFL
      WA = SPEED(1)
      RETURN
200 SPEED(2) = WA
      SPEED(3) = SPEED(1)
      SPEED(1) = 2.0*WA-SPEED(1)
      WEIGHT(2) = WTFL
      WEIGHT(3) = WEIGHT(1)
      WA = SPEED(1)
      RETURN
210 WEIGHT(1) = WTFL
      IF (WTFL.GE.WT) GO TO 140
      IF (WEIGHT(1)-WEIGHT(2)) 230,380,220
220 WEIGHT(3) = WEIGHT(2)
      WEIGHT(2) = WEIGHT(1)
      SPEED(3) = SPEED(2)
      SPEED(2) = SPEED(1)
      SPEED(1) = 2.0*SPEED(2)-SPEED(3)
      WA = SPEED(1)
      RETURN
230 IF (SPEED(3)-SPEED(1)-10.0) 170,170,240
240 IND = 4
245 IF (WEIGHT(3)-WEIGHT(1)) 260,260,250
250 WA = (SPEED(1)+SPEED(2))/2.0
      RETURN

```

```

260 WA = (SPEED(3)+SPEED(2))/2.0
    RETURN
270 IF (SPEED(3)-SPEED(1)-10.0) 170,170,280
280 IF (WTFL-WEIGHT(2)) 320,350,290
290 IF (WA-SPEED(2)) 310,300,300
300 SPEED(1) = SPEED(2)
    SPEED(2) = WA
    WEIGHT(1) = WEIGHT(2)
    WEIGHT(2) = WTFL
    GO TO 245
310 SPEED(3) = SPEED(2)
    SPEED(2) = WA
    WEIGHT(3) = WEIGHT(2)
    WEIGHT(2) = WTFL
    GO TO 245
320 IF (WA-SPEED(2)) 340,330,330
330 WEIGHT(3) = WTFL
    SPEED(3) = WA
    GO TO 245
340 WEIGHT(1) = WTFL
    SPEED(1) = WA
    GO TO 245
350 IND = 5
    IF (WA-SPEED(2)) 380,360,360
360 SPEED(1) = SPEED(2)
    WEIGHT(1) = WEIGHT(2)
    SPEED(2) = (SPEED(1)+SPEED(3))/2.0
    WA = SPEED(2)
    RETURN
370 IND = 4
    WEIGHT(2) = WTFL
    WA = (SPEED(1)+SPEED(2))/2.0
    RETURN
380 IND = 5
390 WEIGHT(3) = WEIGHT(2)
    SPEED(3) = SPEED(2)
    SPEED(2) = (SPEED(1)+SPEED(3))/2.0
    WA = SPEED(2)
    RETURN
1000 FORMAT (/12H FIXED LINE 12,12H, MAX WT = F10.6)
    END

```


SAMPLE OUTPUT

For aid in checking the operation of the program and as an illustration of the results obtained by the program, sample output is included below. The case used here is the same radial gas turbine rotor with splitter blades that was used for the numerical example in reference (2). The first output is from the calling program (Q3D) and consists of the values of variables MOB, H, M, and S. The value of MOB determines whether the meridional solution (FIXED) or the blade-to-blade solution (SPLBLD) or both is desired. In the example below MOB = 2 which gives both solutions. The value of H, M, and S is used to indicate whether the hub, meridional or shroud blade-to-blade solution is desired; the value 1 indicates that particular solution is desired, the value 0 not. In the example H = 1, and M and S = 0.

The data given immediately following "RUN NO. 1 INPUT DATA CARD LISTING", up to the line before "STAG. SPEED OF SOUND AT INLET", are a listing of the data on the input data cards for the meridional solution. This is followed by a list of the maximum computed streamline change at each iteration for the meridional solution. Convergence to the specified accuracy of .001" was obtained after 105 iterations. Then the calculated information is printed out for every 5th streamline. Data at more streamlines will be printed out if the value of NPRT is reduced from 5.

The output for the meridional solution is given in two parts. The first part gives values of DN, Z, R, WA, PRESS, WTR, and RC and the maximum streamline change for iteration No. 100. The second part for iteration No. 107 gives values for ALPHA, RC, SM, BETA, TT, SA, SB, SC, and SD, followed by a repetition of the values of DN, Z, R, WA, PRESS, WTR, and RC. The next output is the "STREAMLINE SPACING ALONG NORMAL", "BLADE COORDINATES", and "INLET ANGLES", which is computed by the meridional program (FIXED) and is used for input to the blade-to-blade program (SPLBLD).

Next is the output from the blade-to-blade program at the hub. The first line is "RUN NO. 2 INPUT DATA CARD LISTING". The next two lines are data from input cards. The following lines up to the 3rd line before "THETA-CALCULATED AND/OR INPUT", are computed by the program, and are the data which would have to be supplied if not furnished by the meridional program. The second line before "THETA-CALCULATED AND/OR INPUT" is data from an input card with a 1 in card column 5, indicating further data is given. A blank card here would indicate no more input for this blade-to-blade solution. The line just before "THETA-CALCULATED AND/OR INPUT" is data from an input card giving information of splitter blade coordinates near the trailing edge that were altered from those computed by the meridional program. The next data are the θ coordinates of the initial equally spaced streamlines at every quasi-orthogonal. This is followed by the stagnation speed of sound, and then a list of the maximum predicted streamline change at each iteration for the blade-to-blade solution. Convergence to the desired accuracy of .001" was obtained in 27 iterations. The desired output is printed out for every 5th streamline at each quasi-orthogonal. Again, data at more streamlines can be obtained with a different value of NPRT.

The mean and shroud blade-to-blade solutions would have also been obtained on the same computer run if we had set M and S equal to 1, and supplied the appropriate input cards.

PUN NO. 1 INPUT DATA CARD LISTING

14	21	7	7	4030.0000	0.6110	11.0000	1.6667	1245.0000	0.000
0	2	0	11	1950.0000	221.5000	0.0247	0.0010	0.2500	
19	5	2	0	2.0000	1.0000	-25.1000	0.0500		
0.7590	0.8100	0.8900	1.0150	1.1200	1.2300	1.3520			
1.5500	1.8030	2.1000	2.3500	2.5500	2.8500	3.1500			
0.	-0.0030	0.0300	0.1400	0.3170	0.5900	0.9430			
1.3550	1.7650	2.1000	2.3500	2.5500	2.8500	3.1500			
3.0100	2.7360	2.5130	2.3230	2.2420	2.1880	2.1600			
2.1470	2.1470	2.1470	2.1470	2.1470	2.1470	2.1470			
3.0100	2.6000	2.2300	1.8220	1.4700	1.1580	0.9230			
0.7970	0.7520	0.7500	0.7500	0.7500	0.7500	0.7500			
-0.	-0.	-0.	-0.	-0.	-0.	-0.			
-0.0012	-0.0044	-0.0201	-0.0567	-0.1248	-0.2339	-0.3966			
-0.6231	-0.8604	-1.0978	-1.3352	-1.5725					
-0.2000	0.	0.2000	0.4000	0.6000	0.8000	1.0000			
1.1000	1.2000	1.4000	1.6000	1.8000	2.0000	2.2000			
2.4000	2.6000	2.8000	3.0000	3.2000					
0.1860	0.1710	0.1560	0.1410	0.1270	0.1160	0.1030			
0.1660	0.150	0.1360	0.1210	0.1065	0.0960	0.0830			
0.1460	0.1310	0.1160	0.1010	0.0870	0.0760	0.0670			
0.1260	0.1110	0.0960	0.0820	0.0670	0.0560	0.0480			
0.1050	0.0910	0.0760	0.0620	0.0480	0.0360	0.0220			
0.0855	0.0710	0.0560	0.0420	0.0280	0.0160	0.0190			
0.0670	0.0510	0.0360	0.0220	0.0080	-0.0050	0.0160			
-0.0040	0.0000	0.8000	1.2000	1.6000	2.0000	2.4000			
0.6000	1.0000	1.4000	1.8000	2.2000	2.6000	3.0000			

STAG. SPEED OF SOUND AT INLET = 2011.55
0.2073

ITERATION NO. 1	MAX. STREAMLINE CHANGE = 0.397387
ITERATION NO. 2	MAX. STREAMLINE CHANGE = 0.364558
ITERATION NO. 3	MAX. STREAMLINE CHANGE = 0.338418
ITERATION NO. 4	MAX. STREAMLINE CHANGE = 0.317072
ITERATION NO. 5	MAX. STREAMLINE CHANGE = 0.294279
ITERATION NO. 6	MAX. STREAMLINE CHANGE = 0.277526
ITERATION NO. 7	MAX. STREAMLINE CHANGE = 0.261206
ITERATION NO. 8	MAX. STREAMLINE CHANGE = 0.245780
ITERATION NO. 9	MAX. STREAMLINE CHANGE = 0.231244
ITERATION NO. 10	MAX. STREAMLINE CHANGE = 0.217469
ITERATION NO. 11	MAX. STREAMLINE CHANGE = 0.204468
ITERATION NO. 12	MAX. STREAMLINE CHANGE = 0.192163
ITERATION NO. 13	MAX. STREAMLINE CHANGE = 0.180295
ITERATION NO. 14	MAX. STREAMLINE CHANGE = 0.169344
ITERATION NO. 15	MAX. STREAMLINE CHANGE = 0.159114
ITERATION NO. 16	MAX. STREAMLINE CHANGE = 0.149380
ITERATION NO. 17	MAX. STREAMLINE CHANGE = 0.140383
ITERATION NO. 18	MAX. STREAMLINE CHANGE = 0.132053
ITERATION NO. 19	MAX. STREAMLINE CHANGE = 0.124201
ITERATION NO. 20	MAX. STREAMLINE CHANGE = 0.116800
ITERATION NO. 21	MAX. STREAMLINE CHANGE = 0.109090
ITERATION NO. 22	MAX. STREAMLINE CHANGE = 0.099879
ITERATION NO. 23	MAX. STREAMLINE CHANGE = 0.093197
ITERATION NO. 24	MAX. STREAMLINE CHANGE = 0.087511
ITERATION NO. 25	MAX. STREAMLINE CHANGE = 0.082277
ITERATION NO. 26	MAX. STREAMLINE CHANGE = 0.077409
ITERATION NO. 27	MAX. STREAMLINE CHANGE = 0.072972
ITERATION NO. 28	MAX. STREAMLINE CHANGE = 0.068739
ITERATION NO. 29	MAX. STREAMLINE CHANGE = 0.064801
ITERATION NO. 30	MAX. STREAMLINE CHANGE = 0.061076

ITERATION NO. 31	MAX. STREAMLINE CHANGE = 0.057551
ITERATION NO. 32	MAX. STREAMLINE CHANGE = 0.054177
ITERATION NO. 33	MAX. STREAMLINE CHANGE = 0.051032
ITERATION NO. 34	MAX. STREAMLINE CHANGE = 0.049114
ITERATION NO. 35	MAX. STREAMLINE CHANGE = 0.045275
ITERATION NO. 36	MAX. STREAMLINE CHANGE = 0.042680
ITERATION NO. 37	MAX. STREAMLINE CHANGE = 0.040203
ITERATION NO. 38	MAX. STREAMLINE CHANGE = 0.037993
ITERATION NO. 39	MAX. STREAMLINE CHANGE = 0.035860
ITERATION NO. 40	MAX. STREAMLINE CHANGE = 0.033894
ITERATION NO. 41	MAX. STREAMLINE CHANGE = 0.031966
ITERATION NO. 42	MAX. STREAMLINE CHANGE = 0.030159
ITERATION NO. 43	MAX. STREAMLINE CHANGE = 0.028456
ITERATION NO. 44	MAX. STREAMLINE CHANGE = 0.026837
ITERATION NO. 45	MAX. STREAMLINE CHANGE = 0.025363
ITERATION NO. 46	MAX. STREAMLINE CHANGE = 0.023967
ITERATION NO. 47	MAX. STREAMLINE CHANGE = 0.022629
ITERATION NO. 48	MAX. STREAMLINE CHANGE = 0.021369
ITERATION NO. 49	MAX. STREAMLINE CHANGE = 0.020179
ITERATION NO. 50	MAX. STREAMLINE CHANGE = 0.019052
ITERATION NO. 51	MAX. STREAMLINE CHANGE = 0.017998
ITERATION NO. 52	MAX. STREAMLINE CHANGE = 0.017001
ITERATION NO. 53	MAX. STREAMLINE CHANGE = 0.016058
ITERATION NO. 54	MAX. STREAMLINE CHANGE = 0.015168
ITERATION NO. 55	MAX. STREAMLINE CHANGE = 0.014330
ITERATION NO. 56	MAX. STREAMLINE CHANGE = 0.013539
ITERATION NO. 57	MAX. STREAMLINE CHANGE = 0.012791
ITERATION NO. 58	MAX. STREAMLINE CHANGE = 0.012087
ITERATION NO. 59	MAX. STREAMLINE CHANGE = 0.011487
ITERATION NO. 60	MAX. STREAMLINE CHANGE = 0.010813
ITERATION NO. 61	MAX. STREAMLINE CHANGE = 0.010206
ITERATION NO. 62	MAX. STREAMLINE CHANGE = 0.009696
ITERATION NO. 63	MAX. STREAMLINE CHANGE = 0.009128
ITERATION NO. 64	MAX. STREAMLINE CHANGE = 0.008669
ITERATION NO. 65	MAX. STREAMLINE CHANGE = 0.008158
ITERATION NO. 66	MAX. STREAMLINE CHANGE = 0.007750
ITERATION NO. 67	MAX. STREAMLINE CHANGE = 0.007291
ITERATION NO. 68	MAX. STREAMLINE CHANGE = 0.006937
ITERATION NO. 69	MAX. STREAMLINE CHANGE = 0.006519
ITERATION NO. 70	MAX. STREAMLINE CHANGE = 0.006204
ITERATION NO. 71	MAX. STREAMLINE CHANGE = 0.005890
ITERATION NO. 72	MAX. STREAMLINE CHANGE = 0.005517
ITERATION NO. 73	MAX. STREAMLINE CHANGE = 0.005248
ITERATION NO. 74	MAX. STREAMLINE CHANGE = 0.004985
ITERATION NO. 75	MAX. STREAMLINE CHANGE = 0.004665
ITERATION NO. 76	MAX. STREAMLINE CHANGE = 0.004424
ITERATION NO. 77	MAX. STREAMLINE CHANGE = 0.004201
ITERATION NO. 78	MAX. STREAMLINE CHANGE = 0.003997
ITERATION NO. 79	MAX. STREAMLINE CHANGE = 0.003801
ITERATION NO. 80	MAX. STREAMLINE CHANGE = 0.003538
ITERATION NO. 81	MAX. STREAMLINE CHANGE = 0.003360
ITERATION NO. 82	MAX. STREAMLINE CHANGE = 0.003189
ITERATION NO. 83	MAX. STREAMLINE CHANGE = 0.003028
ITERATION NO. 84	MAX. STREAMLINE CHANGE = 0.002872
ITERATION NO. 85	MAX. STREAMLINE CHANGE = 0.002728
ITERATION NO. 86	MAX. STREAMLINE CHANGE = 0.002603
ITERATION NO. 87	MAX. STREAMLINE CHANGE = 0.002406
ITERATION NO. 88	MAX. STREAMLINE CHANGE = 0.002284
ITERATION NO. 89	MAX. STREAMLINE CHANGE = 0.002178
ITERATION NO. 90	MAX. STREAMLINE CHANGE = 0.002058

ITERATION NO.	1	2	3	4A	PRESS	WTR	RC
MAX. STREAMLINE CHANGE =	3.010955	377.965088	10.798088	-0.000000	1.137485		
MAX. STREAMLINE CHANGE =	3.001857	269.531698	10.213267	-0.000000	2.938320		
MAX. STREAMLINE CHANGE =	3.001761	254.311989	9.744524	-0.000000	5.758334		
MAX. STREAMLINE CHANGE =	3.001675	249.271402	9.309812	-0.000000	5.339774		
MAX. STREAMLINE CHANGE =	3.001593	242.361450	9.007061	-0.000000	5.142885		
MAX. STREAMLINE CHANGE =	3.001515	237.816582	8.765873	-0.000000	5.527833		
MAX. STREAMLINE CHANGE =	3.001440	229.855871	8.650198	-0.000000	5.238314		
MAX. STREAMLINE CHANGE =	3.001368	229.725634	8.551350	-0.000000	5.541405		
MAX. STREAMLINE CHANGE =	3.001301	326.493009	8.417940	-0.000000	3.392933		
MAX. STREAMLINE CHANGE =	3.001238	518.566555	8.049547	-0.000000	5.541405		
MAX. STREAMLINE CHANGE =	3.001176	686.538170	7.658735	-0.000000	3.392933		
MAX. STREAMLINE CHANGE =	3.001119	645.716760	7.752184	-0.000000	5.541405		
MAX. STREAMLINE CHANGE =	3.001064	679.323357	7.648843	-0.000000	3.392933		
MAX. STREAMLINE CHANGE =	3.001017	687.157875	7.611016	-0.000000	5.541405		
MAX. STREAMLINE CHANGE =	3.000962						
MAX. STREAMLINE CHANGE =	3.010000	315.162716	10.787358	-0.000000	2.365052		
MAX. STREAMLINE CHANGE =	2.541074	294.622234	10.233665	-0.000000	5.269357		
MAX. STREAMLINE CHANGE =	2.327013	305.187008	9.790317	-0.000000	7.964499		
MAX. STREAMLINE CHANGE =	2.021102	324.327397	9.400115	-0.000000	9.593285		
MAX. STREAMLINE CHANGE =	1.808114	345.814606	9.145297	-0.000000	9.377177		
MAX. STREAMLINE CHANGE =	1.543605	369.580849	8.948527	-0.000000	7.553375		
MAX. STREAMLINE CHANGE =	1.522074	340.953114	8.876734	-0.000000	7.755554		
MAX. STREAMLINE CHANGE =	1.429487	364.588326	8.750883	-0.000000	7.777719		
MAX. STREAMLINE CHANGE =	1.353671	440.289177	8.547409	-0.000000	7.559528		
MAX. STREAMLINE CHANGE =	1.275650	616.855354	8.096533	-0.000000	5.274153		
MAX. STREAMLINE CHANGE =	1.222082	792.763168	7.564175	-0.000000	9.123643		
MAX. STREAMLINE CHANGE =	1.202778	717.302895	7.759147	-0.000000	9.123643		
MAX. STREAMLINE CHANGE =	1.192958	727.215691	7.705014	-0.000000	9.907273		
MAX. STREAMLINE CHANGE =	1.191312	730.766541	7.672916	-0.000000	5.384133		
MAX. STREAMLINE CHANGE =	3.010000	325.856609	10.770970	-0.000000	2.497993		
MAX. STREAMLINE CHANGE =	2.577043	325.781162	10.272789	-0.000000	6.155512		
MAX. STREAMLINE CHANGE =	2.402795	365.779358	9.787646	-0.000000	9.455482		
MAX. STREAMLINE CHANGE =	2.154085	411.974716	9.395245	-0.000000	11.392922		
MAX. STREAMLINE CHANGE =	2.003585	448.593491	9.152268	-0.000000	11.555781		
MAX. STREAMLINE CHANGE =	1.993074	440.446377	9.043899	-0.000000	9.863162		
MAX. STREAMLINE CHANGE =	1.812042	437.556976	8.957722	-0.000000	8.772571		
MAX. STREAMLINE CHANGE =	1.745977	444.555134	8.864200	-0.000000	5.153480		
MAX. STREAMLINE CHANGE =	1.690791	513.557175	8.653472	-0.000000	7.182481		
MAX. STREAMLINE CHANGE =	1.626111	694.439308	8.136867	-0.000000	9.176489		
MAX. STREAMLINE CHANGE =	1.574744	937.868149	7.434427	-0.000000	5.495793		

STREAMLINE 16	0.810334	2.550000	1.560569	798.011703	7.742152	-0.000000	2.224055
	0.802349	2.650000	1.552666	793.150475	7.727714	-0.000000	0.535912
	0.800654	3.150000	1.550957	793.904666	7.700324	-0.000000	0.337582
STREAMLINE 17	0.581970	0.582132	3.013000	340.181992	10.748186	-0.000000	2.923375
	0.645643	0.645643	2.708506	369.315697	10.210643	-0.000000	7.375245
	0.739818	0.739818	2.463580	436.510109	9.739882	-0.000000	10.753508
	0.860844	0.860844	2.252895	519.938873	9.282377	-0.000000	15.539882
	0.968276	1.014925	2.160981	543.744606	8.089550	-0.000000	12.117793
	1.064116	1.151556	2.051753	543.444934	8.993367	-0.000000	15.487104
	1.144435	1.302244	2.009515	534.171577	8.947193	-0.000000	10.649172
	1.185000	1.524391	1.959639	522.184326	8.913221	-0.000000	3.755375
	1.190466	1.797410	1.941799	501.572403	8.722108	-0.000000	-1.254417
	1.157251	2.100000	1.905984	762.738274	8.163368	-0.000000	-1.580209
	1.127390	2.500000	1.877456	967.544664	7.352173	-0.000000	5.332144
	1.119212	2.550000	1.869535	870.431419	7.743914	-0.000000	0.731579
	1.115434	2.850000	1.865875	961.875244	7.743157	-0.000000	0.334517
	1.114562	3.150000	1.864989	861.700546	7.714558	-0.000000	0.164197
STREAMLINE 21	0.758788	0.759000	3.013000	357.868290	10.718747	0.	3.256959
	0.824178	0.810000	2.735300	414.276752	10.168045	0.	7.827293
	0.905361	0.890000	2.513000	519.959923	9.617226	0.	12.614864
	1.010889	1.015000	2.328000	651.520416	9.033949	0.	21.203702
	1.114054	1.120000	2.242339	670.325665	8.872091	0.	19.584972
	1.212764	1.230000	2.188000	692.510811	8.737076	0.	25.159187
	1.302942	1.352000	2.160000	639.562263	8.642950	0.	13.785172
	1.364230	1.550000	2.147000	590.028603	8.936932	0.	3.761135
	1.395782	1.803000	2.147000	433.591988	8.770262	0.	0.716506
	1.397322	2.100000	2.147000	828.493217	8.181007	0.	0.212459
	1.396918	2.350000	2.147000	1051.483170	7.342592	0.	-0.051705
	1.396596	2.500000	2.147000	937.041249	7.758596	0.	0.313247
	1.396448	2.850000	2.147000	829.850731	7.759997	0.	-0.332133
	1.396473	3.150000	2.147000	929.614441	7.728097	0.	-1.703173

MAX. STREAMLINE CHANGE = 0.000915

ITERATION NO.106

ITERATION NO.107	ALPHA	RC	SM	BETA	TT	SA	SR	SC	SD
STREAMLINE 1	1.197495	0.	0.	-25.085542	0.067282	-0.035649	2409.414703	0.981628	4.623333
	-92.079852	0.410011	0.781480	-1.355874	0.039464	0.092429	473.735358	2.935831	-14.135967
	-88.196739	0.781480	1.204048	-0.207217	0.102356	0.922232	294.623634	5.684514	-47.998913
	-80.784849	1.204048	1.598044	-0.000190	0.119518	1.898475	272.632027	4.990815	-103.532155
	-69.177340	1.598044	2.012620	-0.315887	0.139582	4.488724	-245.542917	6.793964	152.314234
	-56.547535	2.012620	2.436688	-0.215887	0.135975	4.941300	150.630854	4.387139	-165.170145
	-41.600297	2.436688	2.867575	-0.452844	0.134488	4.387074	173.644947	3.873397	-357.595192
	-28.788874	2.867575	3.276987	-1.674529	0.126025	4.636741	438.544333	0.878905	1201.911514
	-10.733212	3.276987	3.614993	-31.152805	0.119840	2.252521	2199.423676	0.082541	4457.370311
	-2.107958	3.614993	3.864993	-41.573327	0.103499	-4.776226	4187.341248	0.019777	462.843252
	0.231244	3.864993	4.064993	-41.573327	0.035497	-0.961608	5359.006897	-0.019777	1935.435272
	-0.035674	4.064993	4.364993	-41.573327	0.	-7.076388	5350.850281	0.002238	-2403.280823
	0.019123	4.364993	4.664993	-41.573327	0.	-7.076388	5359.884094	-0.000694	2129.153374
	-0.005654	4.664993	4.964993	-41.573327	0.	-7.076388	5357.933105	-0.000661	-1139.172180
	0.006545	4.964993	5.264993	-41.573327	0.	-7.076388	5357.933105	-0.000661	-1139.172180
STREAMLINE 6	2.045052	0.	0.	-25.051566	0.059338	0.111445	2189.869476	1.674544	15.216747
	-86.192426	0.371252	0.695910	-1.993864	0.074753	0.910801	-747.588462	9.183583	35.508580
	-80.034306	0.695910	1.040993	0.300043	0.037284	2.735934	-497.570671	7.479737	161.941804
	-69.908507	1.040993	1.396473	0.077030	0.096899	5.471243	-583.926216	7.880142	403.315497

DN	1	Z	R	4A	PRESS	WTR	RC
STREAMLINE 11							
-42.256077	9.370177	1.323094	-0.036073	0.100518	6.935300	-919.105937	6.400921
-30.959021	7.566375	1.600244	0.130237	0.100520	6.937713	134.557387	3.893339
-21.081983	7.253768	1.870726	-2.419274	0.097436	6.697071	602.144417	7.2581752
-13.777719	7.196640	2.155924	-11.559241	0.092929	3.301640	1369.583198	0.552127
-12.697451	7.254041	2.506793	-28.807953	0.066789	3.301640	3610.050262	0.759087
-16.103229	8.204163	2.868165	-44.319237	0.098858	4.590942	5348.411671	1.153440
-8.436362	9.123643	3.123841	-55.123064	0.062374	-3.941251	6735.892517	0.590598
-3.405184	2.908270	3.324770	-56.870350	0.	-5.712037	6651.931052	-0.339076
-0.843168	0.802779	3.624930	-56.773328	0.	-6.445198	6576.935187	0.004855
0.000064	0.364937	3.924937	-56.713967	0.	-6.588154	6579.024559	0.000007
2.497980	0.	0.	-24.921461	0.051853	0.274004	1633.758885	2.037204
-82.339543	6.365612	0.302235	7.355171	7.355171	1.585778	-1225.057376	117.4246335
-75.612821	0.629759	0.629759	-0.075044	0.075044	4.057786	-1463.916743	8.552033
-64.616480	1.392922	0.925033	-0.339337	0.081331	7.370506	-1821.467529	697.358617
-49.686907	1.1555781	1.143061	0.155729	0.083321	9.157116	-446.663904	1547.261227
-37.561159	1.349809	1.349809	-1.359155	0.082728	7.850189	-462.410981	7.948410
-27.483977	8.683362	1.349809	-5.336945	0.080423	8.181177	823.455223	4.087922
-18.681746	8.723751	1.556557	-16.719355	0.077098	2.265741	2410.478882	-2.766271
-11.522017	1.820095	1.820095	-16.719355	0.077098	2.265741	2410.478882	1515.153819
-8.723751	1.820095	1.820095	-16.719355	0.077098	2.265741	2410.478882	1515.153819
-1.064981	2.437351	-35.307061	-3.557184	0.028252	-3.557184	4660.927278	-0.665317
-12.709867	2.437351	-35.307061	-3.557184	0.028252	-3.557184	4660.927278	-0.665317
-10.64492	2.437351	-35.307061	-3.557184	0.028252	-3.557184	4660.927278	-0.665317
-12.709867	2.437351	-35.307061	-3.557184	0.028252	-3.557184	4660.927278	-0.665317
-8.723751	1.820095	1.820095	-16.719355	0.077098	2.265741	2410.478882	1515.153819
-1.064981	2.437351	-35.307061	-3.557184	0.028252	-3.557184	4660.927278	-0.665317
-12.709867	2.437351	-35.307061	-3.557184	0.028252	-3.557184	4660.927278	-0.665317
-10.64492	2.437351	-35.307061	-3.557184	0.028252	-3.557184	4660.927278	-0.665317
-12.709867	2.437351	-35.307061	-3.557184	0.028252	-3.557184	4660.927278	-0.665317
-8.723751	1.820095	1.820095	-16.719355	0.077098	2.265741	2410.478882	1515.153819
-1.064981	2.437351	-35.307061	-3.557184	0.028252	-3.557184	4660.927278	-0.665317
-12.709867	2.437351	-35.307061	-3.557184	0.028252	-3.557184	4660.927278	-0.665317
-10.64492	2.437351	-35.307061	-3.557184	0.028252	-3.557184	4660.927278	-0.665317
-12.709867	2.437351	-35.307061	-3.557184	0.028252	-3.557184	4660.927278	-0.665317
-8.723751	1.820095	1.820095	-16.719355	0.077098	2.265741	2410.478882	1515.153819

0.	0.	-0.003000	2.600000	269.531698	10.213267	44.381239	2.938333
0.	0.030000	2.230000	2.230000	255.311989	9.745574	17.649894	5.758339
0.	0.140000	1.822000	1.822000	249.271402	9.309812	85.447875	5.339774
0.	0.317000	1.470000	1.470000	242.361450	9.007061	131.817118	5.142884
0.	0.590000	1.155000	1.155000	237.816582	8.765873	197.366289	5.507833
0.	0.943000	0.923000	0.923000	229.855871	8.650198	205.535545	9.230318
0.	1.355000	0.797000	0.797000	229.725344	8.581350	215.551575	4.841405
0.	1.765000	0.742000	0.742000	326.452009	8.417940	284.071049	3.892939
0.	2.100000	0.750000	0.750000	518.568585	8.089547	393.212330	-0.516884
0.	2.350000	0.750000	0.750000	686.538170	7.658735	547.579430	-0.199342
0.	2.550000	0.750000	0.750000	645.716766	7.752184	636.029503	-0.042550
0.	2.850000	0.750000	0.750000	679.323377	7.648843	670.534271	0.799495
0.	3.150000	0.750000	0.750000	687.157875	7.611016	689.271565	0.004745
STREAMLINE 6							
0.	0.201052	3.010000	3.010000	315.162880	10.787358	204.097549	2.765237
0.	0.247549	2.541075	2.541075	294.623375	10.233666	79.501489	5.253315
0.	0.310379	2.327019	2.327019	305.190873	9.790318	74.900695	7.944472
0.	0.395792	2.020113	2.020113	324.333592	9.400119	165.568123	9.593239
0.	0.487955	1.808135	1.808135	345.822739	9.145304	237.262314	9.373257
0.	0.571804	1.643613	1.643613	369.568024	8.948539	297.768276	7.565363
0.	0.631045	1.522127	1.522127	343.942872	8.876746	211.972967	7.205534
0.	1.462364	1.429523	1.429523	364.595916	8.750896	207.946825	2.717501
0.	1.781391	1.353707	1.353707	440.297020	8.547420	233.863977	-1.663452
0.	2.100000	1.275685	1.275685	616.861465	8.096542	373.503254	0.733925
0.	2.350000	1.222112	1.222112	792.771034	7.564170	680.776035	9.123551
0.	2.550000	1.202893	1.202893	717.308334	7.759149	730.896538	3.928151
0.	2.850000	1.192983	1.192983	727.220551	7.705014	716.165359	0.920778
0.	3.150000	1.191034	1.191034	730.775477	7.672917	733.246048	0.394392
STREAMLINE 11							
0.	0.395533	3.012000	3.012000	325.857048	10.770970	224.906893	2.497965
0.	0.457608	2.577351	2.577351	328.782814	10.232790	127.450715	5.453345
0.	0.552826	2.402804	2.402804	365.786694	9.787643	142.700815	9.455879
0.	0.663473	2.154120	2.154120	411.984879	9.395243	256.489874	11.393339
0.	0.770040	2.033609	2.033609	448.606117	9.152268	335.957592	11.516149
0.	0.865545	1.893107	1.893107	440.459671	9.042905	238.306179	8.942204
0.	0.936476	1.812079	1.812079	437.571918	8.957729	235.808084	8.773945
0.	1.492080	1.746016	1.746016	444.565285	8.84217	191.624953	3.153740
0.	1.790574	1.593939	1.593939	513.566574	8.653436	251.132151	-1.957088
0.	2.100000	1.626199	1.626199	694.445938	8.130874	376.590378	-1.764787
0.	2.350000	1.576779	1.576779	903.876777	7.434424	770.723373	8.595582
0.	2.550000	1.550603	1.550603	798.018745	7.742193	830.394569	2.224011
0.	2.850000	1.552693	1.552693	793.158302	7.727709	781.337574	0.537241
0.	3.150000	1.550981	1.550981	793.913995	7.700321	802.454147	3.307735
STREAMLINE 16							
0.	0.582138	3.010000	3.010000	340.182526	10.748185	247.099993	2.833298
0.	0.645654	2.708508	2.708508	369.316677	10.210844	183.061054	7.075131
0.	0.739834	2.453585	2.453585	436.518093	9.733873	220.094879	10.753237
0.	0.845143	2.252994	2.252994	519.947258	9.282370	357.520537	15.541154
0.	1.014942	2.140998	2.140998	543.757790	9.089538	256.899998	12.115574
0.	1.151568	2.051774	2.051774	543.459011	8.993366	268.854904	15.888821
0.	1.302257	2.009539	2.009539	534.182228	8.947193	261.808247	10.543207
0.	1.464462	1.922521	1.922521	522.189231	8.913236	191.402967	3.755932
0.	1.524385	1.949655	1.949655	581.572255	8.722123	276.503910	-1.254530
0.	1.797411	1.941825	1.941825	762.741524	8.163381	404.695511	-1.480363
0.	2.100000	1.907038	1.907038	987.548767	7.352176	849.245132	6.332313
0.	2.350000	1.877478	1.877478	870.436935	7.743912	908.501259	0.731805
0.	2.550000	1.959555	1.959555	870.436935	7.743912	908.501259	0.731805
0.	2.850000	1.855892	1.855892	861.882187	7.743148	847.635330	0.334714
0.	3.150000	1.865004	1.865004	861.705940	7.716552	873.430992	2.164245
STREAMLINE 21							
0.	0.759900	3.010000	3.010000	357.848176	10.718747	268.014072	3.056949

0.824179	0.810000	2.736000	414.274696	10.168049	245.513897	7.827293
0.905361	0.890000	2.513000	519.360502	9.617225	302.936909	12.414864
1.010492	1.015000	2.328000	651.511513	9.033974	256.796584	21.203702
1.114056	1.120000	2.242000	670.213733	8.872125	286.295414	19.584972
1.212764	1.230000	2.198000	652.300496	8.737106	329.763535	25.159147
1.302944	1.352000	2.150000	639.575328	8.842977	313.160950	13.283172
1.364235	1.500000	2.147000	590.221835	8.936949	222.181139	3.376135
1.395786	1.803000	2.147000	643.588181	8.770273	305.830257	-0.715586
1.397328	2.100000	2.147000	828.489792	8.181018	461.813942	0.212459
1.396922	2.350000	2.147000	1051.482574	7.342595	917.599991	-0.061785
1.396794	2.500000	2.147000	937.062500	7.758592	960.334343	0.213247
1.396442	2.850000	2.147000	929.853561	7.755987	915.630478	-0.302939
1.396468	3.150000	2.147000	929.616028	7.728092	941.762290	-0.301473

ITERATION 03.107

MAX. STRAINLINE CHANGE = 0.000871

STREAMLINE SPACING ALONG NORMAL

HUB	MEAN	SHROUD
0.040693	0.037813	0.034168
0.052181	0.040605	0.031432
0.068673	0.042892	0.028822
0.074510	0.044754	0.025841
0.130443	0.045669	0.025764
0.170835	0.045781	0.026383
0.205420	0.048879	0.028604
0.200044	0.052244	0.033175
0.175676	0.055190	0.038419
0.134393	0.060227	0.045457
0.110665	0.063937	0.052036
0.102950	0.065784	0.057418
0.099293	0.066623	0.054367
0.098594	0.066764	0.054241

BLADE COORDINATES

HUB				MEAN				SHROUD				KMX			
I	KMX	KHPI	KMX	I	KMX	KHPI	KMX	I	KMX	KHPI	KMX	I	KMX	KHPI	KMX
-0.5600	-0.2968	-0.2744	-0.0112	-0.5526	-0.2942	-0.2770	-0.0086	-0.5649	-0.2920	-0.2793	-0.0064	-0.5622	-0.2906	-0.2817	-0.0044
-0.5548	-0.3020	-0.2692	-0.0164	-0.5500	-0.2978	-0.2734	-0.0122	-0.5597	-0.2966	-0.2761	-0.0156	-0.5580	-0.2950	-0.2744	-0.0184
-0.5482	-0.3085	-0.2626	-0.0229	-0.5450	-0.3012	-0.2700	-0.0190	-0.5580	-0.2990	-0.2724	-0.0222	-0.5580	-0.3013	-0.2733	-0.0250
-0.5384	-0.3184	-0.2528	-0.0328	-0.5323	-0.3045	-0.2667	-0.0267	-0.5580	-0.3056	-0.2773	-0.0313	-0.5580	-0.3143	-0.2866	-0.0384
-0.5268	-0.3300	-0.2412	-0.0444	-0.5203	-0.3078	-0.2641	-0.0342	-0.5580	-0.3143	-0.2866	-0.0384	-0.5580	-0.3260	-0.2975	-0.0456
-0.5175	-0.3443	-0.2269	-0.0587	-0.5097	-0.3142	-0.2596	-0.0459	-0.5580	-0.3260	-0.2975	-0.0459	-0.5580	-0.3440	-0.3172	-0.0544
-0.4984	-0.3585	-0.2128	-0.0729	-0.5052	-0.3142	-0.2596	-0.0549	-0.5580	-0.3440	-0.3172	-0.0549	-0.5580	-0.3620	-0.3347	-0.0644
-0.5103	-0.3828	-0.2247	-0.0972	-0.5028	-0.3142	-0.2596	-0.0644	-0.5580	-0.3620	-0.3347	-0.0644	-0.5580	-0.3800	-0.3544	-0.0744
-0.5036	-0.4074	-0.3180	-0.1918	-0.5003	-0.3142	-0.2596	-0.0744	-0.5580	-0.3800	-0.3544	-0.0744	-0.5580	-0.4000	-0.3844	-0.0844
-0.8085	-0.6021	-0.5229	-0.3765	-0.5003	-0.3142	-0.2596	-0.0844	-0.5580	-0.4000	-0.3844	-0.0844	-0.5580	-0.4200	-0.4044	-0.0944
-1.0977	-0.8861	-0.8121	-0.6007	-1.1129	-0.8698	-0.8283	-0.5842	-1.1192	-0.8698	-0.8335	-0.5842	-1.1192	-0.8861	-0.8488	-0.6044
-1.3724	-1.0868	-1.0868	-0.8012	-1.3724	-1.0868	-1.0868	-0.8012	-1.3724	-1.0868	-1.0868	-0.8012	-1.3724	-1.0868	-1.0868	-0.8012
-1.7284	-1.4428	-1.4428	-1.1572	-1.7284	-1.4428	-1.4428	-1.1572	-1.7284	-1.4428	-1.4428	-1.1572	-1.7284	-1.4428	-1.4428	-1.1572
-2.0844	-1.7988	-1.7988	-1.5132	-2.0844	-1.7988	-1.7988	-1.5132	-2.0844	-1.7988	-1.7988	-1.5132	-2.0844	-1.7988	-1.7988	-1.5132

INLET ANGLES - HUB -24.56- MEAN -23.38, SHROUD -21.61

RUN NO. 2 INPUT DATA CARD LISTING

22	8	0	2	0.1500	0.0010	0.2000		
0.		3.3100		0.0330				
15	22	8	0	4030.0000	0.0305	11.0000	1.6567	1245.0000
2	2	0	12	1950.0000	221.5000	0.0247	0.0010	0.2500 6941.2549
0	5	2	0	-24.5566	0.0000	0.1500		
-0.5561		-0.5600		-0.5548	-0.5482	-0.5384	-0.5268	-0.5125
-0.4984		-0.5103		-0.6036	-0.8085	-1.0977	-1.3724	-1.7284
-2.0844								
0.0151		-0.0112		-0.0164	-0.0229	-0.0328	-0.0444	-0.0587
-0.0729		-0.0972		-0.1918	-0.3765	-0.6005	-0.9012	-1.1572
-1.5132								
-0.2705		-0.2968		-0.3020	-0.3085	-0.3184	-0.3300	-0.3443
-0.3585								
-0.2705		-0.2744		-0.2692	-0.2626	-0.2528	-0.2412	-0.2269
-0.2128								
0.		0.		-0.0002	0.0025	0.0117	0.0264	0.0492
0.0786		0.1129		0.1471	0.1750	0.1958	0.2125	0.2375
0.2625								
0.2758		0.2508		0.2167	0.1858	0.1518	0.1225	0.0965
0.0769		0.0664		0.0627	0.0625	0.0625	0.0625	0.0625
0.0625								
0.0032		0.0034		0.0043	0.0057	0.0079	0.0109	0.0142
0.0172		0.0167		0.0146	0.0112	0.0092	0.0085	0.0083
0.0082								

1
711-0.33890 811-0.31900 712-0.23630 812-0.28660 812-0.28660

THETA-CALCULATED AND/OR INPUT

-0.5561	-0.5600	-0.5548	-0.5482	-0.5384	-0.5268	-0.5125
-0.4984	-0.5103	-0.6036	-0.8085	-1.0977	-1.3724	-1.7284
-2.0844						
-0.5275	-0.5337	-0.5295	-0.5243	-0.5164	-0.5071	-0.4957
-0.4844	-0.4897	-0.5830	-0.7869	-1.0728	-1.3438	-1.6998
-2.0558						
-0.4990	-0.5074	-0.5042	-0.5003	-0.4944	-0.4874	-0.4789
-0.4704	-0.4690	-0.5624	-0.7653	-1.0479	-1.3153	-1.6713
-2.0273						
-0.4704	-0.4810	-0.4789	-0.4763	-0.4724	-0.4678	-0.4621
-0.4564	-0.4484	-0.5419	-0.7437	-1.0231	-1.2867	-1.6427
-1.9987						
-0.4418	-0.4547	-0.4537	-0.4524	-0.4504	-0.4481	-0.4453
-0.4424	-0.4277	-0.5213	-0.7221	-0.9982	-1.2582	-1.6141
-1.9701						
-0.4133	-0.4284	-0.4284	-0.4284	-0.4284	-0.4284	-0.4284
-0.4284	-0.4071	-0.5007	-0.7005	-0.9737	-1.2296	-1.5856
-1.9416						
-0.3847	-0.4021	-0.4031	-0.4044	-0.4064	-0.4087	-0.4116
-0.4145	-0.3864	-0.4801	-0.6789	-0.9485	-1.2010	-1.5570
-1.9130						
-0.3562	-0.3757	-0.3779	-0.3805	-0.3844	-0.3890	-0.3948
-0.4005	-0.3657	0.4595	-0.6573	-0.9236	-1.1725	-1.5285
-1.8845						
-0.3276	-0.3494	-0.3526	-0.3565	-0.3624	-0.3694	-0.3780
-0.3865	-0.3451	-0.4289	-0.6357	-0.8988	-1.1439	-1.4999
-1.8559						
-0.2990	-0.3231	-0.3273	-0.3325	-0.3404	-0.3497	-0.3612
-0.3725	-0.3244	-0.4143	-0.6141	-0.8739	-1.1154	-1.4713
-1.8273						
-0.2705	-0.2968	-0.3020	-0.3085	-0.3184	-0.3300	-0.3389
-0.3190	-0.3038	-0.3977	-0.5925	-0.8491	-1.0868	-1.4428
-1.7988						

-0.2705	-0.2744	-0.2692	-0.2626	-0.2528	-0.2412	-0.2363
-0.2866	-0.3038	-0.3977	-0.5925	-0.8491	-1.0868	-1.4428
-1.7988						
-0.2419	-0.2481	-0.2439	-0.2387	-0.2308	-0.2215	-0.2101
-0.1988	-0.2831	-0.3771	-0.5709	-0.8242	-1.0582	-1.4142
-1.7702						
-0.2134	-0.2218	-0.2186	-0.2147	-0.2088	-0.2018	-0.1933
-0.1848	-0.2625	-0.3565	-0.5493	-0.7993	-1.0297	-1.3857
-1.7417						
-0.1848	-0.1954	-0.1935	-0.1907	-0.1858	-0.1822	-0.1765
-0.1708	-0.2418	-0.3359	-0.5277	-0.7745	-1.0011	-1.3571
-1.7131						
-0.1562	-0.1691	-0.1681	-0.1668	-0.1648	-0.1625	-0.1597
-0.1568	-0.2212	-0.3153	-0.5061	-0.7496	-0.9726	-1.3285
-1.6845						
-0.1277	-0.1428	-0.1428	-0.1428	-0.1428	-0.1428	-0.1428
-0.1428	-0.2005	-0.2947	-0.4845	-0.7248	-0.9440	-1.3000
-1.6560						
-0.0991	-0.1165	-0.1175	-0.1188	-0.1208	-0.1231	-0.1260
-0.1289	-0.1799	-0.2742	-0.4629	-0.6999	-0.9154	-1.2714
-1.6274						
-0.0706	-0.0901	-0.0923	-0.0949	-0.0988	-0.1034	-0.1092
-0.1149	-0.1592	-0.2536	-0.4413	-0.6750	-0.8969	-1.2429
-1.5989						
-0.0420	-0.0638	-0.0670	-0.0709	-0.0768	-0.0838	-0.0924
-0.1009	-0.1386	-0.2330	-0.4197	-0.6502	-0.8533	-1.2143
-1.5703						
-0.0134	-0.0375	-0.0417	-0.0469	-0.0548	-0.0641	-0.0756
-0.0869	-0.1179	-0.2124	-0.3981	-0.6253	-0.9298	-1.1857
-1.5417						
0.0151	-0.0112	-0.0164	-0.0229	-0.0328	-0.0444	-0.0587
-0.0729	-0.0972	-0.1918	-0.3765	-0.6005	-0.8012	-1.1572
-1.5132						

STAG. SPEED OF SOUND AT INLET = 2011.55

ITERATION NO. 1	MAX. STREAMLINE CHANGE = 0.068444
ITERATION NO. 2	MAX. STREAMLINE CHANGE = 0.058218
ITERATION NO. 3	MAX. STREAMLINE CHANGE = 0.049416
ITERATION NO. 4	MAX. STREAMLINE CHANGE = 0.041949
ITERATION NO. 5	MAX. STREAMLINE CHANGE = 0.035614
ITERATION NO. 6	MAX. STREAMLINE CHANGE = 0.030236
ITERATION NO. 7	MAX. STREAMLINE CHANGE = 0.025670
ITERATION NO. 8	MAX. STREAMLINE CHANGE = 0.021794
ITERATION NO. 9	MAX. STREAMLINE CHANGE = 0.018503
ITERATION NO. 10	MAX. STREAMLINE CHANGE = 0.015709
ITERATION NO. 11	MAX. STREAMLINE CHANGE = 0.013338
ITERATION NO. 12	MAX. STREAMLINE CHANGE = 0.011324
ITERATION NO. 13	MAX. STREAMLINE CHANGE = 0.009614
ITERATION NO. 14	MAX. STREAMLINE CHANGE = 0.008163
ITERATION NO. 15	MAX. STREAMLINE CHANGE = 0.006931
ITERATION NO. 16	MAX. STREAMLINE CHANGE = 0.005885
ITERATION NO. 17	MAX. STREAMLINE CHANGE = 0.004996
ITERATION NO. 18	MAX. STREAMLINE CHANGE = 0.004248
ITERATION NO. 19	MAX. STREAMLINE CHANGE = 0.003601
ITERATION NO. 20	MAX. STREAMLINE CHANGE = 0.003061
ITERATION NO. 21	MAX. STREAMLINE CHANGE = 0.002595
ITERATION NO. 22	MAX. STREAMLINE CHANGE = 0.002207
ITERATION NO. 23	MAX. STREAMLINE CHANGE = 0.001876
ITERATION NO. 24	MAX. STREAMLINE CHANGE = 0.001588

ITERATION NO. 25
 MAX. STREAMLINE CHANGE = 0.001350
 ITERATION NO. 26
 MAX. STREAMLINE CHANGE = 0.001148
 ITERATION NO. 27
 MAX. STREAMLINE CHANGE = 0.000976
 ITERATION NO. 28
 MAX. STREAMLINE CHANGE = 0.000829

ITERATION NO. 29
 QUASI-ORTHOGONAL 1 SM = -0.3000 ALPHA = -89.41 R = 3.3100 Z = 0. DN = 0.0380

THETA	T-CURV	BETA	WA	WTHETA	VTHETA	WM	V	PRS	DENSITY	DTDM	DWDM	SA	S3	DMDT
-0.5561	41.46	-48.57	375.8	-281.8	829.8	246.7	866.3	11.22	0.02271	-4.11	6941.	0.4961	-204.1	-17.7
-0.4148	34.12	-48.57	373.2	-279.8	831.8	247.0	867.7	11.23	0.02271	-4.11	6941.	0.4961	-204.1	-19.0
-0.2726	30.29	-48.57	370.4	-277.7	833.9	245.1	869.2	11.23	0.02272	-4.11	6941.	0.4961	-204.1	-20.4
-0.2726	42.80	-48.57	370.4	-277.7	833.9	245.1	869.2	11.23	0.02272	-4.11	6941.	0.4961	-204.1	-20.4
-0.1293	33.77	-48.57	367.4	-275.5	836.1	243.1	870.8	11.24	0.02272	-4.11	6941.	0.4961	-204.1	-21.9
0.0151	28.91	-48.57	364.1	-273.0	838.6	241.0	872.5	11.24	0.02273	-4.11	6941.	0.4961	-204.1	-23.5

QUASI-ORTHOGONAL 2 SM = 0. ALPHA = -90.72 R = 3.0100 Z = 0. DN = 0.0407

THETA	T-CURV	BETA	WA	WTHETA	VTHETA	WM	V	PRS	DENSITY	DTDM	DWDM	SA	S3	DMDT
-0.5600	-16.89	5.70	290.8	28.9	1059.7	289.3	1379.2	12.82	0.02220	0.40	9107.	-0.0988	261.4	232.8
-0.4339	-13.71	-1.75	288.3	-8.8	1002.1	286.2	1342.7	10.83	0.02221	-0.12	6941.	0.0105	-280.5	-271.7
-0.2967	-8.67	-0.62	252.0	-2.7	1008.1	252.0	1339.2	10.87	0.02226	-0.04	7999.	0.0108	-15.2	-12.5
-0.2744	-26.3	5.97	295.8	30.8	1041.6	294.2	1382.4	10.82	0.02219	0.47	8984.	-0.1034	230.7	270.1
-0.1495	-13.0	-1.94	289.5	-9.8	1001.1	289.4	1342.0	10.82	0.02220	-0.14	6784.	0.0336	-319.8	-310.3
-0.0112	-7.16	-0.88	246.5	-3.8	1007.1	246.5	1336.8	10.89	0.02227	-0.06	7798.	0.0154	-65.4	-51.5

QUASI-ORTHOGONAL 3 SM = 0.4100 ALPHA = -89.60 R = 2.6000 Z = -0.0030 DN = 0.0522

THETA	T-CURV	BETA	WA	WTHETA	VTHETA	WM	V	PRS	DENSITY	DTDM	DWDM	SA	S3	DMDT
-0.5548	0.26	1.97	531.9	16.3	391.5	531.5	1037.9	9.73	0.02081	0.05	-7379.	-0.0344	-2303.2	-2321.5
-0.4751	11.56	-6.86	359.8	-43.0	130.2	357.3	903.8	10.08	0.02126	-0.56	-892.	0.1185	-1925.1	-1862.4
-0.3020	-0.20	-2.16	28.8	-1.1	872.1	28.8	872.6	10.38	0.02163	-0.20	-2101.	0.0370	-2199.4	-2193.3
-0.2692	0.34	1.94	530.2	16.0	891.1	529.9	1036.8	9.74	0.02082	0.05	-2541.	-0.0331	-2295.0	-2312.9
-0.1892	11.04	-6.66	358.0	-41.5	831.7	355.5	904.5	10.09	0.02126	-0.54	-808.	0.1151	-1908.0	-1855.8
-0.0184	-0.22	-2.14	31.4	-1.2	872.0	31.4	872.6	10.38	0.02163	-0.19	-2017.	0.0374	-2181.3	-2180.1

QUASI-ORTHOGONAL 4 SM = 0.7815 ALPHA = -80.63 R = 2.2300 Z = 0.3300 DN = 0.0687

THETA	T-CURV	BETA	WA	WTHETA	VTHETA	WM	V	PRS	DENSITY	DTDM	DWDM	SA	S3	DMDT
-0.5482	0.12	2.63	404.9	18.6	757.5	404.4	867.5	9.53	0.02052	0.18	767.	-0.0453	-1333.9	-1352.7
-0.4580	-5.28	6.51	302.4	34.3	783.2	300.5	838.8	9.69	0.02073	0.61	1878.	-0.1111	-1121.5	-1155.1
-0.3085	-0.19	-2.51	112.5	-5.1	743.8	112.3	752.2	9.86	0.02096	-0.14	500.	0.0449	-1383.5	-1378.4
-0.2626	0.19	2.72	407.1	19.3	768.2	406.6	869.2	9.52	0.02052	0.18	655.	-0.0467	-1354.6	-1373.6
-0.1727	-8.86	6.49	303.5	34.3	783.2	301.5	839.5	9.69	0.02073	0.61	1761.	-0.1109	-1143.3	-1175.9
-0.0229	-0.16	-2.60	110.6	-5.0	743.9	110.5	752.1	9.86	0.02096	-0.14	460.	0.0447	-1391.0	-1386.1

QUASI-ORTHOGONAL 5 SM = 1.2040 ALPHA = -69.23 R = 1.8720 Z = 0.1400 DN = 0.0945

THETA	T-CURV	BETA	WA	WTHETA	VTHETA	WM	V	PRS	DENSITY	DTDM	DWDM	SA	S3	DMDT
-0.5384	0.22	2.77	402.8	19.5	631.4	402.3	748.7	9.09	0.01993	0.16	-238.	-0.0451	-1178.9	-1197.1
-0.4553	2.97	1.05	296.0	5.4	617.3	296.0	684.6	9.25	0.02015	0.12	-956.	-0.0172	-1289.1	-1294.2
-0.3184	-0.13	-2.71	124.4	-5.9	606.0	124.3	618.6	9.41	0.02035	-0.17	-258.	0.0442	-1182.1	-1175.6
-0.2520	-0.04	2.50	399.4	17.4	629.3	399.0	745.2	9.10	0.01994	0.16	77.	-0.0408	-1131.5	-1147.8

[illegible]

THETA	T-CURV	BETA	WA	WTHETA	VTWETA	WM	V	PRS	DENSITY	DTDM	DMCM	SA	S3	DMOT
-0.8085	-11.00	-34.56	695.2	-394.4	-142.5	572.5	590.0	7.65	0.01784	-1.21	-10037.	-0.0019	-515.0	
-0.7095	-8.74	-32.31	644.9	-344.7	-92.9	545.0	552.9	7.70	0.01802	-1.02	-9395.	-0.0018	-495.8	
-0.6058	-8.76	-30.84	595.4	-305.2	-53.3	511.2	514.0	7.90	0.01819	-1.05	-8551.	-0.0018	-459.2	
-0.6058	-8.67	-30.87	595.4	-305.5	-53.6	511.0	513.8	7.90	0.01819	-1.26	-8554.	-0.0018	-458.2	
-0.4955	-8.38	-30.09	546.9	-274.2	-22.4	473.2	473.7	8.01	0.01834	-9.27	-7807.	-0.0018	-421.4	
-0.3765	-9.09	-29.93	498.4	-248.8	3.1	432.0	432.0	8.11	0.01847	-1.37	-7341.	-0.0017	-396.7	

QUASI-ORTHOGONAL 12 SM = 3.8650 ALPHA = -0.06 R = 0.7500 Z = 2.3500 DN = 0.1107

THETA	T-CURV	BETA	WA	WTHETA	VTWETA	WM	V	PRS	DENSITY	DTDM	DMCM	SA	S3	DMOT
-1.0977	-2.88	-45.28	742.2	-527.4	-275.5	522.2	590.4	7.51	0.01762	-4.59	-89.	0.0005	-4.3	
-0.5698	-3.48	-42.23	735.3	-494.2	-242.3	544.4	595.9	7.53	0.01765	-14.52	-2181.	0.0005	-101.3	
-0.8460	-3.30	-39.51	718.2	-456.9	-205.1	554.1	590.8	7.57	0.01772	-4.77	-3523.	0.0005	-170.3	
-0.8460	-3.32	-39.50	718.2	-456.9	-205.0	554.2	592.9	7.57	0.01772	-4.72	-3514.	0.0005	-169.5	
-0.7536	-2.22	-37.30	694.8	-421.0	-189.2	537.7	578.0	7.64	0.01781	-12.19	-4202.	0.0005	-209.0	
-0.6005	-0.17	-35.82	667.0	-390.9	-139.0	541.5	559.2	7.71	0.01790	-4.48	-4414.	0.0005	-223.8	

QUASI-ORTHOGONAL 13 SM = 4.0650 ALPHA = 0.02 R = 0.7500 Z = 2.5500 DN = 0.1030

THETA	T-CURV	BETA	WA	WTHETA	VTWETA	WM	V	PRS	DENSITY	DTDM	DMCM	SA	S3	DMOT
-1.3724	3.85	-44.60	669.9	-470.3	-218.4	475.9	524.6	7.69	0.01787	-11.02	3400.	-0.0002	147.0	
-1.2234	1.44	-42.64	687.0	-474.0	-222.1	497.2	544.5	7.65	0.01781	-15.26	1784.	-0.0002	80.7	
-1.0794	-0.71	-42.29	693.7	-466.7	-214.9	513.1	556.3	7.63	0.01778	-9.55	284.	-0.0002	13.1	
-1.0794	-0.71	-42.29	693.7	-466.7	-214.9	513.1	556.2	7.63	0.01778	-9.56	277.	-0.0002	12.9	
-0.9391	-2.68	-40.73	671.3	-451.0	-199.2	523.8	560.4	7.63	0.01779	-13.78	-938.	-0.0002	44.3	
-0.8012	-4.37	-39.08	682.4	-430.1	-178.2	529.7	558.9	7.66	0.01783	-9.21	-1671.	-0.0002	-80.9	

QUASI-ORTHOGONAL 14 SM = 4.3650 ALPHA = -0.01 R = 0.7500 Z = 2.8500 DN = 0.0993

THETA	T-CURV	BETA	WA	WTHETA	VTWETA	WM	V	PRS	DENSITY	DTDM	DMCM	SA	S3	DMOT
-1.7284	-1.10	-40.69	721.1	-470.2	-218.3	546.8	588.8	7.54	0.01764	-16.16	-944.	0.0000	-44.7	
-1.5870	0.36	-41.67	717.2	-475.0	-223.1	537.4	581.9	7.55	0.01765	-14.14	-323.	0.0000	-15.2	
-1.4456	0.98	-42.01	715.6	-478.9	-227.1	531.7	578.2	7.55	0.01766	-13.19	-245.	0.0000	-11.4	
-1.4456	0.98	-42.01	715.6	-478.9	-227.1	531.7	578.2	7.55	0.01766	-13.19	-242.	0.0000	-11.2	
-1.3020	1.07	-42.33	713.0	-480.1	-228.2	527.1	574.4	7.56	0.01767	-14.57	-599.	0.0000	-27.7	
-1.1572	0.79	-42.44	707.0	-477.0	-225.2	521.0	568.3	7.58	0.01769	-11.55	-1250.	0.0000	-57.5	

QUASI-ORTHOGONAL 15 SM = 4.6650 ALPHA = 0.00 R = 0.7500 Z = 3.1500 DN = 0.0986

THETA	T-CURV	BETA	WA	WTHETA	VTWETA	WM	V	PRS	DENSITY	DTDM	DMCM	SA	S3	DMOT
-2.0844	-0.49	-42.43	688.8	-464.3	-212.9	509.4	551.2	7.61	0.01772	-15.78	1145.	-0.0000	52.8	
-1.9373	0.19	-40.87	698.3	-456.9	-205.0	528.1	566.5	7.58	0.01769	-13.84	1562.	-0.0000	73.9	
-1.7942	0.54	-40.31	709.4	-458.8	-207.0	541.0	579.2	7.55	0.01764	-14.56	1642.	-0.0000	78.3	
-1.7942	0.54	-40.31	709.4	-458.8	-207.0	541.0	579.2	7.55	0.01764	-14.56	1639.	-0.0000	78.2	
-1.6533	0.60	-40.44	719.8	-466.9	-215.0	547.8	588.5	7.52	0.01760	-13.64	1431.	-0.0000	68.1	
-1.5132	0.43	-41.04	727.9	-477.9	-226.0	549.0	593.0	7.50	0.0177	-12.99	939.	-0.0000	44.3	

ITERATION NO. 29

MAX. STREAMLINE CHANGE = 0.000705

REFERENCES

1. Katsanis, Theodore: Use of Arbitrary Quasi-Orthogonals for Calculating Flow Distribution in a Turbomachine. ASME Paper no. 65-WA/GTP-2, 1965.
2. Katsanis, Theodore: Use of Arbitrary Quasi-Orthogonals for Calculating Flow Distribution in the Meridional Plane of a Turbomachine. NASA TN D-2546, 1965.
3. Katsanis, Theodore: Use of Arbitrary Quasi-Orthogonals for Calculating Flow Distribution on a Blade-to-Blade Surface in a Turbomachine. NASA TN D-2809, 1965.
4. Hamming, R. W.: Numerical Methods for Scientists and Engineers. McGraw-Hill Book, Co., Inc., 1962, p. 314.